AD-A157 724 MERSUREMENT OF RF (RADIO FREQUENCY) FIELDS ASSOCIATED $1 / 2$ HITH ISM (INDUSTRIA. (U) OHIO UNIY RTHENS RVIDNICS ENGIMEERING CEMTER J D HICKUM ET AL. MAY 85
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Systems Engineering Service Washington, D.C. 20591

## Measurement of RF Fields Associated With ISM Equipment as it Relates to Aeronautical Services

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May 1985
Final Report

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Described are the RF field measurements of four Industrial, Scientific and Medical devices to characterize the fundamental and 4 th harmonic radiation from these devices according to FCC Part 18 and CISPR Publication 11 and ilA. The effects of the 4 th harmonic radiation is considered with respect to ILS localizer receiver susceptibility. The testing was performed at an open field test site with measurements made on the ground and at elevation angles from 45 to 75 degrees. Additionally, an aircraft equipped with calibrated antennas was flown over the ISM device to determine the RF fields radiated overhead at the th harmonic of the fundamental operating frequency. The four ISM devices consisted of one with 25 kW power output, two with 2 kW power output, and one at 3 kW power output.
Results indicate that RF fields (at ILS localize frequencies) of 20 to 40 dB above $\operatorname{FCC}$ limits can exist as a result of ISM equipment 4 th harmonic emissions in the vicinity of ISM equipment. Co-channel emissions from ISM devices on localizer frequencies can produce CDI deviations in excess of 5 at the missed approach point for certain locations of ISM equipment.

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Industrial, Scientific and Medical, ISM Airborne measurements, RF Field Measurements, ILS Localizer.

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## English/Metric Conversion Factors

## Length

| Front | Cm | m | Km | in | 4 | 3 mi | nmi |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| Cm | 1 | 0.01 | $1 \times 10^{-5}$ | 0.3937 | 0.0328 | $6.21 \times 10^{-6}$ | 5.39×10.6 |
| m | 100 | 1 | 0.001 | 39.37 | 3.281 | 0.0006 | 0.0005 |
| Km | 100.000 | 1000 | 1 | 39370 | 3281 | 0.6214 | 0.5395 |
| in | 2.540 | 0.0254 | $2.54 \times 10^{.5}$ | 1 | 0.0833 | $1.58 \times 10^{-5}$ | $1.37 \times 10^{.5}$ |
| ft | 30.48 | 0.3048 | $3.05 \times 10^{-4}$ | 12 | 1 | $1.89 \times 10^{-4}$ | $1.64 \times 10^{-4}$ |
| Smi | 160.900 | 1609 | 1.609 | 63360 | 5280 |  | 0.8688 |
| nmi | 185.200 | 1852 | 1.852 | 72930 | 6076 | 1.151 |  |

Area

|  | $\mathrm{Cm}^{2}$ | $\mathrm{m}^{2}$ | Km ${ }^{2}$ | in ${ }^{2}$ | $\mathrm{ft}^{2}$ | $S \mathrm{mi}^{2}$ | $n m i^{2}$ |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| $\mathrm{Cm}^{2}$ | 1 | 0.0001 | $1 \times 10^{10}$ | 0.1550 | 0.0011 | $3.86 \times 10^{.11}$ | $5.11 \times 10^{.11}$ |
| $\mathrm{m}^{2}$ | 10.000 | 1 | $1 \times 10^{-6}$ | 1550 | 10.76 | $3.86 \times 10^{-7}$ | $5.11 \times 10^{-7}$ |
| Km ${ }^{2}$ | $1 \times 10^{10}$ | $1 \times 10^{6}$ | 1 | $1.55 \times 10^{9}$ | $1.08 \times 10^{7}$ | 0.3861 | 0.2914 |
| $\mathrm{in}^{2}$ | 6.452 | 0.0006 | $6.45 \times 10^{-10}$ | 1 | 0.0069 | $2.49 \times 10^{-10}$ | $1.88 \times 10^{.10}$ |
| $4 t^{2}$ | 929.0 | 0.0929 | $9.29 \times 10^{-8}$ | 144 | 1 | $3.59 \times 10^{-8}$ | $2.71 \times 10^{.8}$ |
| $\mathrm{S} \mathrm{mi}{ }^{2}$ | $2.59 \times 10^{10}$ | $2.59 \times 10^{6}$ | 2.590 | $4.01 \times 10^{9}$ | $2.79 \times 10^{7}$ | 1 | 0.7548 |
| $n m i^{2}$ | $3.43 \times 10^{10}$ | $3.43 \times 10^{6}$ | 3.432 | $5.31 \times 10^{9}$ | $3.70 \times 10^{7}$ | 1.325 | 1 |

Volume

|  | $\mathrm{Cm}^{3}$ | Liter | $\mathrm{m}^{3}$ | in ${ }^{3}$ | $4^{3}$ | yd ${ }^{3}$ | 1102 | $1{ }_{1} \mathrm{pr}$ | 41 at | gal |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| $\mathrm{Cm}^{3}$ | 1 | 0.001 | $1 \times 10^{-6}$ | 0.0610 | $3.53 \times 10^{-5}$ | $1.31 \times 10^{-6}$ | 0.0338 | 0.0021 | 0.0010 | 0.0002 |
| liter | 1000 | 1 | 0.001 | 61.02 | 0.0353 | 0.0013 | 33.81 | 2.113 | 1.057 | 0.2642 |
| $\mathrm{m}^{2}$ | $7 \times 10^{6}$ | 1000 | 1 | 61.000 | 35.31 | 1.308 | 33.800 | 2113 | 1057 | 264.2 |
| in ${ }^{3}$ | 16.39 | 0.0163 | $1.64 \times 10^{-5}$ | 1 | 0.0006 | $2.14 \times 10^{-5}$ | 0.5541 | 0.0346 | 2113 | 0.0043 |
| $4{ }^{3}$ | 28.300 | 28.32 | 0.0283 | 1728 | 1 | 0.0370 | 957.5 | 59.84 | 0.0173 | 7.481 |
| Yo ${ }^{3}$ | 765.000 | 764.5 | 0.7646 | 46700 | 27 | 1 | 25900 | 1616 | 807.9 | 202.0 |
| 1102 | 29.57 | 0.2957 | $2.96 \times 10^{-5}$ | 1.805 | 0.0010 | $3.87 \times 10^{-5}$ | 1 | 0.0625 | 0.0312 | 0.0078 |
| fi pt | 473.2 | 0.4732 | 0.0005 | 28.88 | 0.0167 | 0.0006 | 16 | 1 | 0.5000 | 0.1250 |
| fl at | 946.3 | 0.9463 | 0.0009 | 57.75 | 0.0334 | 0.0012 | 32 | 2 | 1 | 0.2500 |
| gal | 3785 | 3.785 | 0.0038 | 231.0 | 0.1337 | 0.0050 | 128 | 8 | 4 | 1 |

Mass

| To |  |  |  |  |  |
| :--- | :--- | :--- | :--- | :--- | :--- |
| Fromin | $g$ | Kg | 02 | 10 | ton |
| 0 | 1 | 0.001 | 0.0353 | 0.0022 | $1.10 \times 10^{.6}$ |
| Kg | 1000 | 1 | 35.27 | 2.205 | 0.0011 |
| 02 | 28.35 | 0.0283 | 1 | 0.0625 | $3.12 \times 10^{-5}$ |
| 10 | 453.6 | 0.4536 | 16 | 1 | 0.0005 |
| ron | 907.000 | 907.2 | 32.000 | 2000 | 1 |

Temperature
${ }^{\circ} \mathrm{C}=5 / 9\left({ }^{\circ} \mathrm{F}=32\right)^{\circ}$
${ }^{\circ} \mathrm{F}=8 / 3\left({ }^{\circ} \mathrm{C}\right)+32$
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This report details the procedures, messurements, anslysis, and recommendations of a measurements program which was designed to determine the radio frequency (RF) fields of the fundamental and 4 th harmonics of devices classed as industrial, scientific, and medical. This equipment is licensed to operate from $26.96-27.28 \mathrm{MHz}$; and the emission of harmonics is regulated by the United States under the Federal Communications Commission (FCC) Part 18 of the Rules and Regulations.

The 4 th harmonics of this equipment fall within the frequency allocation of the aeronautical instrument landing system (ILS) band. Therefore, the FAA is interested in what real RF fields exist over and around any industrial, scientific and medical (ISM) device and what RF fields are capable of causing serious interference to aeronautical users. The FAA's interest is in obtaining actual measured results in order to substantiate requests made to the FCC to increase or decrease emissions standards for certification of ISM devices. This report presents data comparing the RF fields measured, based on FCC and Comite International Special Des Perturbations Radioelectriques (CISPR) procedures, to the RF fields measured by an aircraft flying over the ISM device at various elevation angles.

Four ISM devices were tested for this report. One device with an RF power output of 25 kW , two devices at 2 kW , and one device at 3 kW were tested at the Elite Electronic Engineering open field test site in Waterman, Illinois. The class of ISM devices tested was dielectric sealers used to seal vinyl and other similar materials. The load used for these tests was silicone, in order to obtain a longer dwell time for ease in making the RF measurements.

The equipment selected for these tests was chosen as a representative range of devices currently used in the industry with power outputs in the range of $2-25 \mathrm{~kW}$. Additionally, the ISM devices were new equipment.

Tests were performed in three categories. The first was that RF emission tests were to be made according to FCC and CISPR procedures as if the equipment were to be certified for use. Second, a set of tests was made such that an antenna could be placed at various elevation angles to measure any radiation occurring at vertical angles. Third, a set of measurements was made using an aircraft equipped with calibrated antennas and flown over the ISM device to determine the presence of any significant vertical lobes of RF radiation on the 4 th harmonic of the operating frequency.

Additionally, based on previous studies performed to determine the localizer receiver susceptability of various receivers, comments and recommendations were made to indicate the ability to provide co-channel interference protection from ISM devices to ILS localizer facilities. Additional comments were made regarding the feasibility and cost of performing such measurements using an aircraft and the quality of the measured data vs. the cost.

The measurements, analysis, and recommendations presented in this report are all derived from the four ISM devices tested. The best procedure, method, and equipment available were used consistent with good engineering practice. As with most engineering programs, there is always one more test or refinement possible; this one is no exception. There are still more measurements that desirably could be made to evaluate the suitability of FCC vs. CISPR measurement methods. Nevertheless, this report will present answers to those questions and will state that some others must still be asked.
II. CONCLUSIONS AND RECOMMENDATIONS

## A. CISPR and FCC Testing.

With regard to the ground-based measured data, all four of the ISM devices tested at the open field test aite at Waterman, Illinois, passed the FCC radiated-emissions tests. Two ISM devices passed the CISPR radiated emissions tests. The airborne tests indicate that a different situation exists. None of the ISM equipment could pass either the FCC or CISPR emissions standards. Airborne test data indicate aignificantly higher field strengths than the ground-based emissions measurements. This appears to be due to $R F$ absorption for low elevation angles and the E-field boundary conditions for receiving antennas close to a ground plane.

In considering that the measured $R P$ fialds above the ISM device can be 20 to 40 dB higher than the $R F$ Eields measured on the ground, it is possible for the ISM signal to be 2.8 dB higher than the $91 \mathrm{~dB} \mu \mathrm{~V} / \mathrm{m}$ RF field measured from the ILS localizer at Ohio University. This is based on the minimum measured air-to-ground $R P$ field difference indicated by Table 2 of 16.9 dB higher than measurements made on the ground. If this value is added to the FCC maximum allowable field at 200 feet over the ISM device ( $76.9 \mathrm{~dB} \mu \mathrm{~V} / \mathrm{m}$ ) the result is $93.8 \mathrm{~dB} \mu \mathrm{~V} / \mathrm{m}$. The measured field strength of the localizer at Ohio Uaiversity is $91 \mathrm{~dB} \mu \mathrm{~V} / \mathrm{m}$ as explained above. This difference ( 93.8 - 91.0 ) indicates that the ISM co-channel interference is 2.8 dB above the ILS localizer signal.

The equivalent CISPR comparison produces an interference signal level of $58.6 \mathrm{~dB} \mu \mathrm{~V} / \mathrm{m}$ which results in a localizer-to-interference signal ratio of 32.4 dB .

In all of the above FCC enissions-related considerations, there appears to be no ability to protect the aeronautical user for certain conditions of ISM placement in the service volume of the localizer.

## B. Ground vs. Alrborne Measurements.

Based on the measurement data from the ground-based FCC tests and the airborne tests performed, it is clear that the $R P$ fields existing at vertical angles surrounding an ISM device are substantially higher than the RF fields measured on the ground. The specific difference amounts to measured RF fields between 20 and 40 dB higher than those measured in tests on the ground. This indicates that the current FCC measurement methods are not adequate to protect aeronautical users for certain locations of ISM equipment near localizer facilities.

Based on the equipment tested, some of the devices exhibited higher RF emissions when shields were in place than when shields were removed, at certain frequencies. Additionally, the absolute ip fields radiated at 109 MHz were significantly higher than at 27 Mif for cartain devices. This indicates the need for careful design of $2 P$ ohielding for these devices.

## C. Difficulty and Expense of Airborne Measurements.

The cost of making airborne $R F$ field measurements may not be significant depending on the type of ISM equipment being measured and the location of the equipment to be measured. There are alternatives to making airborne measurements, but these methods provide less complete data relating to the presence and levels of RP fields existing above the ISM equipment. A device comparable to the Clark tower could be used to determine the fields that exist at higher elevation angles, but this method does not provide measurement capability directly over the measured device. If the ISM device is being measured at a site that employs a turntable, using the Clark tower-type device is relatively easy, since the tower can be positioned and the device under test can be rotated on the turntable to make azimuth measurements.

If the device to be tested is located at an operational site, the problem of making these measurements is more significant using the Clark tower-type device. In order to make the measurements, the tower must be moved for each measurement, which is a very time-consuming activity. In this case it may be more cost effective to make the measurements from an aircraft. Most aircraft are already equipped. with VHF antennas that can make the necessary measurement of the 4 th harmonic of the ISM fundamental frequency, and methods do exist that allow calibration of the antenne. It is estimated that the airborne survey, including calibration of the antenna, could be completed with as little as 2.75 hours of flight time. For a single-engine aircraft capable of this operation the total costs of renting an aircraft, including a pilot and engineering labor, would be approximately $\$ 500$ for the complete flight test. This assumes that the necessary receiving equipment is already available. This is not an unusual criterion since the receiving equipment is already required for the ground test procedures. To perform the same number of azimuth measurements using the Clark tower device and estimating 2 hours per messurement total using 2 people at $\$ 20 /$ hour, the labor costs would equate to $\$ 1440$. Additionally, the measurements would take 4.5 chronological days to complete; whereas the flight test data would take less than a day.

It appears that the cost of making the filght measurements is offset by the higher total cost of using a ground-based test device such as the Clark tower. The additional benefit of using the aircraft is that more complete measurements can be made of the RF fields that exist above the ISM device in a shorter time span. If ground based measurement procedures are improved so that adequate prediction of RP fields existing over the equipment can be made, then the need for airborne measurements could be eliminated.

The data collection system configuration (shown in Figure i) consists of a Heath H 89 computer that controls several peripheral devices to collect and record relevant data. These data are the RF E-field amplitude, frequency, the aircraft position, and time of measurement. The H 89 is a complete functional computer that supports a console screen, console keyboard, multiple disk drives, and three RS-232 ports. In addition, FORTH is available for use on the H89. The use of FORTH has resulted in a reduced software development time as compared to assembler with faster execution time compared to BASIC.

To measure the RF interference levels, an Electro-Metrics EMC-25 interference analyzer is incorporated into the system. The EMC-25 is designed for use as the major component of interference analysis systems from 14 kHz to 1 GHz . The receiver is tunable in 15 frequency bands for the range specified and is capable of measuring signal levels from $0 \mathrm{~dB} \mu \mathrm{~V}$ to $120 \mathrm{~dB} \mu \mathrm{~V}$ within $\pm 1.5 \mathrm{~dB} \mu \mathrm{~V}$ at frequencies above $25 \mathrm{MHz}(-20 \mathrm{~dB} \mu \mathrm{~V}$ to 100 $\mathrm{dB} \mu \mathrm{V}$ below 25 MHz ).

Signals provided by the EMC-25 to indicate received signal amplitudes and frequency are dc voltage levels of 0 to +1.5 V nominal. The dc voltage signal for the amplitude is derived from the meter terminal voltage and therefore is an indication of the meter deflection, while the frequency signal is a measure of the tuner setting. In addition to the above signals, there are four binary data lines encoded as a hexadecimal digit that indicates the frequency band number, and seven binary data lines from the attenuator switch. Each data line from the attenuator switcin indicates that a particular attenuator setting has been selected. These seven data lines are encoded by an 8 to 3 line encoder to give a 3-bit octal representation of the attenuator switch position. The EMC-25 also contains a rechargeable battery pack as a power source that will provide enough power for the unft to operate approximately 12 hours between charges. This is an important consideration when operating in a small airplane.

A Serial Lab Products SL-803-A Intelifgent Remote Serial $1 / 0$ unit is used to convert the analog signals from the EMC- 25 into ASCII characters and to make avallable upon request all EMC- 25 signals on a RS- 232 data comaunications link. The SL-803-A was chosen for its wide range of capabilities and for its ease of application. Up to 16 channels of analog data and eight digital input lines may be used. This exceeds the requirement for two channels for a/d conversion and seven digital input lines. The SL-803-A is controlled by characters sent over the RS- 232 line, and it is transparent to any transmission until it detects an ASCII character that has been selected by the user as its control character. Then it reads the subsequent ASCII codes and acts according to the designed command convention. Among the programmable modes of the unit are enabling of specified channels and the selection of either $\pm 2 \mathrm{~V}$ or $\pm 10 \mathrm{~V}$ a/d conversion.

For the RF field measurement to be useful in determining the propagation pattern, the position of each measurement must be recorded. $A$ Motorola Miniranger with telemetry data link is used to measure the
AMPLTWDE FRERUENCY STATUS
distance from a ground point to the airplane while the altitude and airplane heading are manually read from the navigation equipment in the airplane. In performing the data collection maneuvers, the pilot flies in a straight line at a constant altitude directly over the test site. When this is done, the position in space at every point can be calculated from the recorded altitude, magnetic bearing, and Miniranger distance.

The Miniranger provides a measurement of distance between the two Miniranger transponders accurate to $\mathbf{~} 2$ meters and outputs the computation of the range in ASCII characters from the base unit. The Miniranger system data link is a transparent two-way commication link which is used in this system to transmit ASCII characters between the SL-803-A in the airplane and the 489 computer on the ground. For this system the Miniranger transponder will be in the airplane and the base station on the ground with H89 computer and ADPI Byte Bucket tape drive. The SL-803-A, located in the airplane, communicates with the 489 computer by sending and receiving characters over the Miniranger telemetry data link.

Airborne data collection for the tests at Waterman, Illinois, was conduct.ed using the system described here except that the position was recorded using a TI9900 Loran-C receiver and the $H 89$ computer and Byte Bucket tape drive were located the airplane. For these tests the aircraft position was determined by recording the position information from the Loran-C receiver, while collecting data and then calculating the distance from the ISM unit using the position of the ISM unit measured by the Loran-C receiver.

A system clock is also kept so that the time of each measurement can be recorded with the other data. The time of day is useful in data reduction by providing evidence of data collection interruption. The clock is a software counter that keeps time via interrupts provided by the H89.

Data collected by the equipment is stored on magnetic tape by the Analog and Digital Peripherals, Inc. (ADPI) Byte Bucket digital cassette tape player/recorder. The Byte Bucket is a cassette tape drive that can be controlled by the system computer by commands sent on the RS-232 data link. The Byte Bucket uses digital cassette tapes capable of storing up to 230,000 bytes of data per side. This translates into roughly 13,000 sample points per tape.

The data transfer between peripheral devices is controlled by a routine running on the $H 89$ computer. While performing the data collection, the routine runs in a continuous loop that inputs data from the three sources and stores it on tape. The routine also creates a display on the computer's CRT to give the operator an indication of data contents, and checks for input from the console keyboard to accept user commands. User commands are software limited to a predefined set of input that controls when data collection and data storage are enabled. Figure 2 is a photograph of the airborne data collection system used in the Waterman, Illinois, tests.


Figure 2. RF Field Strength Data Collection Equipment as Installed in N8238C for Waterman, IL. Flights

The system for measuring the distance from the unit under test to the aitplane was changed after the Waterman data collection because the present system using the Miniranger provides improved accuracy, is less susceptible to operator erfor, and provides a direct neasuremant of the range. This new system with the Miniranger provides a mathod for measuring signal levels in space that is easy to operate and provides accurate range and signal strength data.

## IV. WATERMAN DATA COLLECTION FLIGHTS

Airborne data collection was conducted at Waterman, Illinois for four pleces of ISM equipment (herein referred to as Machines $A, B, C, \& D$ ). These data collection flights were performed with the ISM oriented so that the maximum lobe of radiation (as detected with ground equipment) coincided with the flight path of the airplane. Also, for Machines B, $C$, and D data collection flights were conducted with the ISM equipment oriented for flight paths at 60 degrees to either side of the maximum lobe. These procedures were consistent with those used in the ground-based measurements using the Clark tower. Three of the Machines (A, C, and D) were tested both with RF shielding on and off to study the effects of shielding while Machine $B$ was tested only with shields on.

Calibration data for equipment, antennas, and cables are indicated in Table 1. For all airborne data this calibration of antennas on the aircraft is appropriate.

## A. Analysis of Airborne Data.

Data collected at Waterman, Illinois, were reduced using the Ohio University IBM 370 computer system and plots of each data run were created. These plots are Figures A-1 - A-21 in Appendix A. The plots show the measured E-field in absolute $\mathrm{dB} \mu \mathrm{V} / \mathrm{m}$ on the ordinate versus the horizontal distance from the test site on the abscissa (refer to Figure 3 for example). The horizontal distance is the distance from a point on the ground directly beneath the airplane to the location of the ISM equipment, and the distance is shown as positive for points north of the test site and negative for points south. The horizontal distance was used to create plots rather than the slant range distance to avoid discontinuities in the graph which would result from the slant range distance ambiguity as the airplane passed over the test site. (The slant range is never less than the aircraft altitude.)

At the top of each plot is a description of the test conditions. This description identifies the machine and indicates the machine setup parameters. Shown on the data plots as dashed lines are the FCC and CISPR limits for this frequency band, calculated by extrapolating the E-field limits from their specified test distance to the distance of concern using the free space decay factor of 2.0 as follows:

$$
E(R)=E_{1 \text { mit }_{t}}\left(\frac{D_{\text {limit }}}{R}\right)^{2.0}
$$

where
$E(R)=E-f i e l d$ limit at distance $R(\mu V / m)$
$\mathrm{E}_{\text {limit }}=$ specified FCC or CISPR E-field limit ( $\mu \mathrm{V} / \mathrm{m}$ )
$D_{\text {imit }}=$ distance at which $E_{\text {1imit }}$ is specified
$R=$ distance of concern

## EMI CALIBRATION DATA <br> February 21, 1984

Biconical AntennaAntenna factor $=16.4 \mathrm{~dB}$ @ 27 MHzAntenna factor = $13.1 \mathrm{~dB} @ 109 \mathrm{MHz}$Source: Three antenna method calibration. Sept. 9, 1983
Bent dipole antenna on Saratoga N8238CAntenna factor $=53.4 \mathrm{~dB}$ @ 27 MHzAntenna factor $=13.1 \mathrm{~dB} @ 109 \mathrm{MHz}$Source: Calibration versus biconical antenna usingsubstitution. Nov. 7, 1983
27 MHz antenna on Saratoga N8238C
Antenna factor $=9 \mathrm{~dB}$ © 27 MHz
Source: Data collected on January 3, 1984
Cables
EMI Cable A (35 feet)
-0.7 dB @ 27 MHz
$-1.2 \mathrm{~dB} @ 109 \mathrm{MHz}$
EMI Cable B (80 feet)
-1.6 dB @ 27 MHz-3.2 dB @ 109 MHz
EMI Cable C (5 feet)
-0.2 dB @ 27 MHz-0.4 dB @ 109 MHz
Source: All cables calibrated Sept. 12, 1983
Dual directional coupler - HP778D serial no. $1144 a 04704$
27 MHz - both ports -32.6 dB109 MHz - both ports -22.0 dB
NOTE: Antenna factor is the value added to the measured fleld in $d B \mu V$ to obtain absolute field strength in $d B_{\mu} V / m$.


Figure 3. Flight Data Machine B, $2 \mathrm{~kW}, 152 \mathrm{M}$ Altitude $0^{\circ}$ Azimuth, RFI Shields in Place

The distance $R$ takes into account the altitude; that is, $R$ is equal to the slant range distance from the ISM equipment to the airplane. The plotted FCC and CISPR limits provide reference points that ease comparison of plots as well as show relevance between actual measured data and maximum permissible levels.

The ISM machines tested at Waterman, Illinois, all exhibited some degree of vertical lobing directly above the unit. The plots of Machine A (Figures A-1 - A-5) show that this machine emitted a relatively low level radiation directly overhead with a uniform higher level at elevation angles to either side of overhead. The plots of this machine's performance show levels as much as 35 dB greater than the FCC limits when some shielding was removed, compared to signal strengths of 6 dB maximum above limits when all shielding was installed properly.

Machine B plots (Figures A-6 - A-11) show that this piece of equipment had some very narrow vertical lobes directly above the unit with uniform signal levels to either side. The lobes of radiation above the unit were as much as 24 dB above FCC limits; whereas radiation to the sides was always within 10 dB of limits.

Machines $C$ and $D$ were the game machine except for the $R F$ power generated; Machine $C$ generated 3 kW of $R F$ power and $D$ produced 2 kW . Neither of these had any significant radiation levels overhead. The largest signal levels detected were about 7 dB above FCC limits with shields off (Figure $A-16$ ) and 5 dB above limits with all shielding in place (figure A-12). The plots of Machine $C$ demonstrate the effects of shielding for this unit. Comparison of Machine $C$ plots where only the shielding is different show that the shielding suppresses the RF signal levels by about 3-7 dB (Figures A-12 to A-17). However, Machine A showed signal levels as much as 15-20 dB higher with shields off as compared to those measured when all shields were in place. This indicates that the shields for Machine A (the 25 kW unit) had a much greater effect on the radiation levels than did the shields on Machine $C$ (a 3 kW unit). This may be due to the design of the shields since there is a lesser need for shielding on the smaller units. Shielding for larger units would naturally be more carefully designed.

The E-field values shown in Table 2 represent the measured field strengths extrapolated to one mile for easy comparison with FCC limits. The ground-based data are those measured by Elite Electronics Engineering Company (under subcontract) using FCC procedures for ISM equipment certification. The airborne data were obtained by evaluating the plots of Figures A-1 - A-22 to find the average difference between the plotted data and the FCC limits. This average difference was taken from a section of the plot that was not directly above the unit. This criterion results in the evaluation of the plots at points where the field is fairly uniform and so represents conditions which would be encountered when flying near one of these units (if flying directly overhead, the signal level could change significantly, either lower or higher). Generally, the points used to generate this table were at a horizontal distance of between -500 and -1500 meters as indicated on the figures. To maintain consistency with the conditions of ground-based measurements, only those data collected with all shielding in place were considered.
table 2. ground vs. airborne data comparison at one mile $\mathrm{dB} \mu \mathrm{V} / \mathrm{m}$

NOTE: PCC LIMIT $=20 \mathrm{~dB} \mu \mathrm{~V} / \mathrm{m}$

## Machine A

| Azimuth | Airborne data <br> $(\mathrm{dB} \mu \mathrm{V} / \mathrm{m})$ | Ground Data <br> $(\mathrm{dB} \mu \mathrm{V} / \mathrm{m})$ | EAG $_{\mathrm{AG}}$ <br> $(\mathrm{dB})$ |
| :---: | :---: | :---: | :---: |
| 180 | +25.0 | -16.4 | 41.4 |

Machine B

| Azimuth | Airborne data <br> $(\mathrm{dB} \mu \mathrm{V} / \mathrm{m})$ | Ground Data <br> $(\mathrm{dB} \mu \mathrm{V} / \mathrm{m})$ | ${ }^{E_{\text {AG }}}$ |
| :---: | :---: | :---: | :---: |
| 0 | 28.0 | +11.1 | $(\mathrm{~dB})$ |
| 240 | 27.0 | +5.9 | 16.9 |
| 300 | 26.0 | +5.3 | 21.1 |
|  |  |  | 20.7 |

Machine C

| Azimuth | Alrborne data <br> $(\mathrm{dB} \mu \mathrm{V} / \mathrm{m})$ | Ground Data <br> $(\mathrm{dB} \mu \mathrm{V} / \mathrm{m})$ | EAG $_{\text {AG }}$ <br> $(\mathrm{dB})$ |
| :---: | :---: | :---: | :---: |
| 20 | +6.0 | -19.5 | 25.5 |
| 260 | +12.0 | -15.9 | 27.9 |
| 320 | +25.0 | -15.0 | 40.0 |

Machine D

| Azimuth | Airborne data <br> $(\mathrm{dB} \mu \mathrm{V} / \mathrm{m})$ | Ground Data <br> $(\mathrm{dB} \mu \mathrm{V} / \mathrm{m})$ | $E_{\mathrm{AG}}$ <br> $(\mathrm{dB})$ |
| ---: | :---: | :---: | :---: |
| 20 | +15.0 | -20.8 | 35.8 |
| 60 | +13.0 | -23.3 | 36.3 |
| 200 | +25.0 | -17.9 | 42.9 |
| 260 | +21.0 | -17.2 | 38.2 |
| 320 | +22.0 | -14.5 | 36.5 |

$E_{A G}=$ Airborne field strength ( $\mathrm{dB} \mu \mathrm{V} / \pi$ ) - Ground field strength ( $\mathrm{dB} \mu \mathrm{V} / \mathrm{m}$ )

In all cases shown in Table 2, the airborne data are much higher than the ground measurements. This difference ranges from 16.9 dB for Machine $B$ at 0 degree to 42.9 dB for Machine D at 200 degrees. Data for Machines B and $D$ show that the difference between ground-based and airborne measurements was relatively constant with respect to azimuth for these two machines. Por Machine $B$ the alrborne measurements ranged from 16.9 to 21.1 dB above ground-based measurements, and airborne measurements for Machine $D$ ranged from 35.8 to 42.9 dB above ground-based measurements (Machine $B$ was tested at three different azimuths and Machine D was tested at five). This seems to indicate that the lobing patterns measured on the ground also exist in the air but with different magnitudes.

The plots of data collected at Waterman, Illinois, exhibit a great deal of consistency conceming the detection of vertical lobing. Every plot shows some amount of lobing at points directly above the unit and a more uniform field at lower elevation angles. The plots indicate that these machines emit a somewhat uniform field with respect to both elevation angle and azimuth (the elevation angles in the plots are always greater than 2.9 degrees). The only lobing with respect to elevation angle is seen directly above the unit. This is similar to the lobing seen from a dipole antenna caused by interaction with the ground plane (see Figure 4). This figure indicates the relative field strength seen by an aircraft making a level pass at 500 feet over the $R F$ source placed 7 feet above the ground [l]. Due to the complex nature of the radiation from ISM equipment, it is expected that a more complex interference pattern would be observed for RF fields directly over the ISM equipment. Machines B, $C$, and $D$ were tested at different azimuths and each displayed a general uniformity of signal levels. Machine $D$ was tested at five different azimuths and, in each case, the received signal was within 5 dB of the FCC limits; however, Machine $C$ did display a significant null at 20 degrees.

Based on the data collected for the four ISM units at Waterman, Illinois, it is seen that the determination of the siznal levels in space produced by a plece of ISM equipment can be measured accurately by flying over the site. The resolution of the data collection system is sufficient to detect most lobing that is present. Additionally, data from the Waterman flight tests seem to indicate that there are no extremely sharf lubes of high level radiation. Since the signal levels measured in the airborne tests were consistently much larger than those measured 0.1 the ground, it seems likely that airborne measurements of the ISM interference signals provide a more accurate measure of the field strengths at high angles than do the ground-based measurements.


Figure 4. Theoretical RF Field Interference Pattern Seen by Aircraft Making Level Poss at 500 ft . with Rf Source at 7 ft . Above Ground

## V. FCC AND CISPR RADIATED EMISSIONS MEASUREMENTS

## A. Test Procedures and Sample Calculations.

1. Open Field Measurements. Measurements were performed at 20 degree increments by turning the units on an air table. Measurements were taken at the fundamental frequency and at all harmonics through the 10 th harmonic. These data were extrapolated to equivalent readings at 1 mile by using a field decay exponent of 1.95 . This decay factor was determined by actual measurement at ground level.

All measurements were performed with the dielectric sealer in a continuous mode of operation (l-minute operation) with a silicon load between the plates. This was done for ease of measurement.

These units were tested at Elite Electronic Engineering Company's Waterman, Illinois, test site (EQU/6810 4-3-0 Elite Engineering Waterman).
2. Distance Correction Calculations. The field intensity limit imposed by the FCC Rules and Regulations is 10 microvolts per meter at 1 mile. Since the data cannot always be taken at 1 mile and since the field intensity from the item is often too weak to be measured at greater distances, especially in the presence of other noise, these data were taken at some closer distance and the field intensity was extrapolated to 1 mile using equation 1. See FCC "Rules and Regulations," Volume II, Part 18, Subpart $D$, para. 18.107 (c).

The propagation decay constant is determined by plotting measured field strength in $d B \mu V$ vs. distance in feet and then drawing an average curve through these points. The slope of this curve is the measured decay constant n. For an example, see Appendix B.

With a measured decay constant $n$, the correction to a distance of 1 mile from any distance $D$ takes the form:

$$
\begin{equation*}
L_{2}=L_{1}(5280 / D)^{-n} \tag{1}
\end{equation*}
$$

$L_{2}=$ Field intensity at 5280 ft .
$L_{1}=$ Measured field intensity at distance $D$
a = Measured decay constant

All data recorded on the data sheets were corrected to equivalent readings at 1 mile. The distance correction factor to convert from 200 feet to 1 wile reduced to -55.4 dB .

The test specification also requires a plot of the equivalent field intensity pattern at 1000 feet to be plotted. The data taken at the fundamental frequency at each azimuth were corrected to equivalent readings at 1000 feet to provide the necessary levels to compose the pattern. See Pigure 6 for an example.

To facilitate the computations which involve antenna factors, calibration factors, and distance factors, the field intensity is first computed in $\mathrm{dB} \mu \mathrm{V} / \mathrm{m}$ and then converted to $\mu \mathrm{V} / \mathrm{m}$ for comparison to the limits.

To obtain the field intensity at a standard distance, the following factors (in dB) are added:

Meter Reading: Obtained from the field intensity meter

## +Antenna Pactor: Supplied by manufacturer of antenna to convert voltage measured at antenna terminals to equivalent voits/meter field intensity

+Distance Correction Factor: Explained above motal in $\mathrm{dB} \mu \mathrm{V} / \mathrm{m}$

This total is converted to $\mu \mathrm{V} / \mathrm{m}$ using the well-known anti-log conversion.

$$
\begin{aligned}
& E(\mu V / m)=10^{\left[\frac{E(d B \mu V / m)}{20}\right]} \\
& \text { B. CISPR vs. FCC Measurement Procedures. }
\end{aligned}
$$

The significant difference in the FCC and CISPR measurement procedures is the distances that the measurements are specified $[2,3,4]$. Since the RF radiation from the ISM devices measured at Waterman, Illinois, was CW, there is no difference in the effective field strengths for CISPR or FCC. The significantly lower CISPR limits seem to be an attempt to account for the fact that when making measurements using an antenna relatively close to the ground, the actual RF field will be higher than indicated for elevation angles above the horizon. Since CISPR specifies measurements at 30 and 100 meters and uses lower radiated limits, the effect at higher elevation angles is that the allowable RF field strength will better represent the line-of-sight $R F$ fields that will exist. The measurements made according to FCC specifications on the ground and extrapolated to 1 mile may be significantly lower than the fields that exist along a direct ine from the unit under test to an aircraft 500 feet or more above the local terrain. It may appear that the CISPR specifications seem to be overly conservative, but they may better protect the aeronautical user since this radiation measurement procedure can better represent the actual launched RF energy when the effect of placing the sensing antenna relatively close to the ground is considered.

This issue of the adoption of CISPR vs. FCC radiation limits is very controversial and needs significant attention. The initial data measurements presented by this report point to the need for additionsl RF radiation measurement procedures for ISM equipment based on FCC limits on interference to ILS localizer facilities.

1. FCC and CISPR ISM Equipment Description. During the open field testing at Waterman, Illinois, four pieces of ISM equipment were tested with the following power output ratings:

| MODEL A | 25 | kW | OUTPUT |
| :--- | ---: | :--- | :--- |
| MODEL B | 2 | kW | OUTPUT |
| MODEL C | 3 | kW | OUTPUT |
| MODEL D | 2 | kW | OUTPPUT |

All of the ground measurement data sheets which include RF field measurements through the l0th harmonic are included in Appendix $B$. The data included here are the radiation pattern measurements at 1000 feet, indicating the shape of the radiation pattern for both the fundamental operating frequency and the 4 th harmonic. The data to generate these plots are derived directly from the ground measurement data sheets contained in Appendix B.

The ground measurement equipment placement for the FCC and CISPR measurements is shown graphically in Figure 5. The biconical antenna used for the ground measurements was placed, for most tests, 200 feet from the ISM device to be measured. The ISM device was set up on the turntable in the building with the position of the operator considered as 0 degree azimuth. After each measurement was made, the ISM device was rotated to the next azimuth angle on the turatable to be measured. In this manner the complete FCC and CISPR emissions tests were made for the device. These results then provided the horizontal lobe of maximum radiation to be considered in the Clark tower and airborne testing.

The spectrum analyzer, its computer and printer, were operated from the instrumentation van by Elite Electronic Engineering Company personnel. This is definitely the most efficient method to make these measurements. The turntable speeds up the positioning of the equipment and the computercontrolled spectrum analyzer speeds up the data-taking and recording. Once the equipment is set up the actual ground testing can be performed in less than an hour on a specific ISM device.
2. FCC and CISPR Emissions Measurements Results. Figures 6 through 13 are the polar plots of the radiation patterns of each of the four ISM devices at the fundamental and 4 th harmonic of the fundamental operating frequency. These data indicate that all of the ISM devices are within the FCC specification for allowable emissions on the 4 th harmonic of the operating frequency. The emissions limit, except for fundamental, extrapolated from 1 mile to the 1000 -foot position is $257 \mu \mathrm{~V} / \mathrm{m}$. This extrapolation was performed using the decay exponent determined by actual ground measurement. A plot of the decay exponent measurement is included in the data for each device contained in Appendix B. The CISPR limit extrapolated to 1000 feet in a similar way produces a limit of $5.4 \mu \mathrm{~V} / \mathrm{m}$. With this limit in mind only Models $C$ and $D, F i g u r e s ~ L l$ and 13 , pass the radiated emissions tests for CISPR. Models A and B, Figures 7 and 9 , exceed CISPR limits for radiated emissions at the 4 th harmonic. This can be seen by referring to the plots for the emissions patterns at the 4 th harmonic. Also of particular note is that there is a considerable amount of correlation between the pattern at 27 MHz and at 108 MHz for ISM Model A, Figures 6 and 7. Prominent RF radiation peaks correlate well between the patterns at the two frequencies. This does not occur when comparing the patterns with any of the other ISM devices, Figures 8 thru 13 . It is not clear why only one of the ISM devices produces a pattern correlation. As was expected, the radiation patterns are quite complex.

LSM/IMD
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TEST

Figure 5. Diagram of Ground, Clark Tower, and Flight Measurement Equipment Setup at Waterman, Illinois


Figure 6. Fundamental Field Intensity Pattern at 1000 ft . for Model A, 25 kW


Figure 7. 4th Harmonic Field Intensity Pattern at 1000 ft . for Model A, 25 kW


Figure 8. F:andamental Field Intensity Pattern at 1000 ft . for Model 8, 2 kW


Figure 9. 4th Harmonic Field Intensity Pattern at 1000 ft . for Model B, 2 kW


Figure 10. Fundamental Field Intensity Portern at 1150 ft . for Model $\mathrm{C}, 3 \mathrm{~kW}$


Figure 11. 4th Harmonic Field Intensity Pattern at 1000 ft . for Model C, 3 kW


Figure 12. Fundamental Field Intensity Pattern ot 1000 ft . for Model D, 2 kW


Figure 13. 4th Harmonic Field Intensity Pattern at 1000 ft . for Model D, 2 kW

All of the results reported in this section are for the equipment operating as per manufacturer's specifications with all radio frequency interference (RFI) shielding in place.

During the course of comparing the ground measurements and the airborne measurements two effects were observed. First, in all of the airborne measurements, as the alrcraft passed directly over the equipment under test, a pattern of nulls and peaks was observed. This was due mainly to the interferometer pattern that was a result of the interaction of the RF source of the ISM equipment interacting with the apparent image source produced by the effects of the ground plane. As the apparent height of the radiating source was located at specific fractions or multiples of the wavelength, an interference pattern of nulls and peaks was formed. This was caused by the differences in the effective paths that the RP energy took to appear at the aircraft antenna position. If the path lengths differ by exactly $1 / 2$ wavelength, the two waves will cancel; and conversely, if the path lengths are 1 wavelength different, then the two waves will add. Therefore, it is easy to understand that for certain geometries the RF energy will appear to produce peaks and nulls. Examining the geometry of the area directly above the ISM equipment, it can be seen that the radiation source and its image are more likely to form these interference patterns since the wave path length differences are greater at this point than when the aircraft is at lower elevation angles. In liont of this, it is clear that in the areas near vertical above the ISM device, the fields can have significant peak-to-peak excursions, but these are true fields and need to be considered when flights over ISM equipment are possible.

The second effect involved a much more subtle consideration but was certainly more significant. Differences in the ground measured data as compared to the airborne measured data for angles greater than about 5-10 degrees up to almost 90 degrees were found. As was determined by previous measurements, the airborne data were some 20 to 40 dB above ground measured data. This may be due to the fact that the radiation measured by the ground tests may be in error of the actual RF fields because at the 100 MHz frequency range the earth conductivity may appear as a lossy dielectric, and the $R F$ fields are attenuated when the recelving antenna is relatively close to the ground. Additionally, considering the antenna patterns of horizontally polarized antennas, it can also be seen that at low elevation angles there is very little $R F$ radfation. This is due to the requirements to satisfy the e-field boundary conditions for horizontally polarized waves. This effect is not the case for vertically polarized waves, but the effects of the ground as a lossy dielectric will generally be of greater importance here. Therefore, it is necessary that these effects be considered when applying procedures used by the FCC and CISPR to make the ground measurements.

## VI. CLARK TOWER OPEN-FIELD TEST PROCEDURES

In order to determine (from the ground) the presence of higher-angle radiation from the ISM unit under test (UUT), a device capable of hoisting an antenna from heights of about 20 feet to 70 feet was used. The tower with the antenna on top was raised to various heights so that the RFfid could be sampled. The tower was placed close to the building containing the UUT so that elevation angles up to approximately 75 degrees could be measured.

## A. Test Equipment.

The test equipment used during this series of tests consists of the following:

1. HP 8568 spectrum analyzer SN 1818 A 00258 Ca 4-9-83
2. HP 9825 computer $S N 1541 A 00350$
3. HP 2631B line printer SN 2002A00184
4. EMCO biconical antenna SN 2171
5. Clark tower pneumatic antenna positioning equipment
B. Procedures.

The Clark tower with the Electro-Mechanics Company (EMCO) biconical antenna mounted on top was positioned 15.75 feet from the center of the turntable used to turn the equipment under test. The tower was positioned at 90 degrees from the direction that the ground RF measurements were made. When tower measurements of the equipment were made, the azimuth indicated In the Clark tower measurements data was the same as the ground measurements data since the turntable was positioned without the 90 -degree offset in azimuth.

The Clark tower base was not at the same level as the equipment under test and therefore the Clark tower height is not the same as the vertical separation of the equipment under test and the antenna on the clark tower. The difference between the base of the Clark tower and the base of the equipment under test was 4 feet. All of the data plots for the Clark tower take this distance difference into account.

Operation of the equipment under test was essentially the same as that in the airborne and ground testing. The ISM equipment was turned on and the RF field measurements were made with the tower at a specific height. The measurements with the Clark tower were made at heights above the base of the tower of 20 feet, 30 feet, 40 feet, 50 feet, and 60 feet, with the azimuth corresponding to the measured maximum $R F$ field from the ground measurements. Aiso, measurements were made 60 degrees to either side of the maximum RF field. Taking into account the difference in the heights of the bases of the equipment under test and the Clark tower, the measured elevation angles correspond to 46 degrees, 59 degrees, 66 degrees, 71 degrees, and 74 degrees. Refer to Figure 5 which indicates the position of Clark tower relative to equipment under test.

The RF measurement device was the $H P 8568$ spectrum analyzer along with the Elitc cable plus the OU 80-foot cable. The data printouts from the Elite spectrum analyzer did not account for the EMCO biconical antenna nor the 80 -foot $O U$ cabie. These values were added to the measured values shown on the Elite data measurement sheets. The values for the EMCO biconical and cables are indicated in Table 1 for 27 MHz and 108 MHz .

The graphic data for the Clark tower measurements were produced by extrapolating the data measured to a common distance of 1000 ft . to allow easy interpretation. This was done by the following method. Using the ground derived decay exponent, the distance correction was determined from the following equation:

$$
F 2=P 1+20 \log \left[\frac{1000}{d}\right]^{-n}
$$

where:

```
Fl = fleld intensity at slant range d in dB\muV/m
F2 = field intensity at range 1000 ft. in dB\muV/m
    n = measured decay exponent
```

The distance $d$ is the distance from the equipment under test to the biconical antenna on the Clark tower.

For example, the slant range from the equipment under test to the biconical on the Clark tower for a tower height of 40 ft . is:

$$
\begin{aligned}
& d=\sqrt{\operatorname{sep}^{2}+(40-\text { delth })^{2}} \\
& d=39.3 \mathrm{ft}
\end{aligned}
$$

where:
sep $=15.75 \mathrm{ft}$. center of curntable to center of Clark tower
delth $=4.0 \mathrm{ft}$. differential in UUT and Clark tower bases
The ground measured decay factor was 1.95. Solving for the RF field at 1000 ft. produces the following result for a measured RF field of 70 $\mathrm{dB} \mu \mathrm{V} / \mathrm{m}$ at the biconical antenna:

$$
F 2=70 \mathrm{~dB} \mu \mathrm{~V} / \mathrm{m}+20 \log \left[\frac{1000}{39.3}\right]^{-1.95}
$$

$\mathrm{P} 2=15.2 \mathrm{~dB} \mu \mathrm{~V} / \mathrm{m}$ or $5.74 \mu \mathrm{~V} / \mathrm{m}$ at 2000 ft.
The following devices were tested using the Clark tower at the open field test site at Waterman, Illinois.
Model $A=25 \mathrm{~kW}$ ISM Device
Model $B=2 \mathrm{~kW}$ ISM Device
Model $C=3 \mathrm{~kW}$ ISM Device
Model $D=2 \mathrm{~kW}$ ISM Device

Models $C$ and $D$ are the same ISM hardware with a different operating RF power output level. Figures 14 - 21 are the graphic representation of the Clark tower data normalized at 1000 ft. range. All ISM equipment, except Model B, have RF field data for the equipment operating with $R F$ shielding intact as per manufacturer's specifications in addition to data with specific RFI shielding removed. In all cases, the shielding is more effective at 27 MHz than at the fourth harmonic of the ISM operating frequency. In Figures 14 and 15 three sets of points are plotted corresponding to all RFI shields on, die table shields removed, and die table and oscillator shields removed. For these configurations some additional explanation is necessary. The device configured with shields removed refers to all RFI shielding surrounding the die table that have been removed along with the cosmetic metal panels surrounding the RF generation unit. The configuration described as "oscillator shields removed" indicates that all RFI shields surrounding the die table have been removed along with the metal closure walls of the master oscillator/power amplifier unit inside the RF power generating unit. In this configuration the cosmetic enclosure panels of the RF power generating unit are in place. This was done to simulate a configuration that might result from maintenance personnel not replacing all of the ISM device RFI shielding after performing maintenance on the unit.

As indicated in Figure 14, the ISM equipment is radiating less energy at 27 MHz with the oscillator shields removed than when all manufacturer's shields are in place. This indicates, to some extent, the differences in the ability of the ISM equipment to launch RF energy based on the device shielding configuration. Figure 15 indicates that at 108 MHz , having all shields in place except die table shields produces no real difference in launched RF energy, but the configuration with the oscillator shields removed has a substantial effect on the launched RF energy at 108 MHz . This is exactly the opposite with the same unit at 27 MHz , where the launched $R F$ energy is lower with the oscillator shields removed than with all of the RFI shields in place.

The limited data of Figures 14 and 15 tend to indicate the presence of lobing in the vertical direction. This can be seen in the dip in the data of Figure 14 at about 60 degrees to the horizon. Also notice the rise in signal level above 60 degrees. This indicates that the unit under test may be radiating a lobe straight up above the unit. This same effect has been indicated in some of the airborne data plots. This tendency of the signal levels to increase for increasing elevarion angles is present in all of the Clark tower measurements at both the fundamental and the 4 th harmonic of the operating frequency. This vertical lobing effect is also indicated in the airborne measurements and is therefore not necessarily a function of the measurement procedures used for the Clark tower measurements.

Observed differences need to be pointed out regarding the measurement of two of the models with the Clark tower and airborne methods. First, the


Figure 14. Fundamental Clark Tower Data for Model A, $25 \mathrm{~kW}, 180^{\circ}$ Azimuth


Figure 15. 4th Harmonic Clark Tower Data for Model A, $25 \mathrm{~kW}, 180^{\circ}$ Azimuth


Figure 16. Fundamental Clark Tower Data for Model B, $2 \mathrm{~kW}, 240^{\circ}$ Azimuth


Figure 17. 4th Harmonic Clark Tower Data for Modal B, $2 \mathrm{~kW}, 240^{\circ}$ Azimuth


Figure 18. Fundamental Clark Tower Data for Model C, $3 \mathrm{~kW}, 320^{\circ}$ Azimuth


Figure 19. 4th Harmonic Clark Tower Dato for Model C, $3 \mathrm{~kW}, 320^{\circ}$ Azimuth


Figure 20. Fundamental Clark Tower Data for Model D, $2 \mathrm{~kW}, 320^{\circ}$ Azimuth


Figure 21 . 4th Harmonic Clark Tower Data for Modal D, $2 \mathrm{~kW}, 320^{\circ}$ Azimuth
azimuth chosen for the Clark tower measuraments for Model A was 180 degrees. This does not colncide with the ground-based lobe of maximum $R P$ radiation. The maximum lobe from the ground-based test reported by Elite was 200 degrees. As indicated on the graphic data for the ISM device Model $A$, the Clark tower and airborne measuraments were made at 180 degrees.

The second situation requiring clarification was that the measurements made on October 13, 1983, (for ISM device Model B for the Clark tower and the airborne measurements) do not reflect the same device tested at Waterman, Illinois. This is due to the fact that the 2 kW device, Model B, did not pass the FCC emissions limits for allowable field intensity at 1 mile. This unit was retested by Elite on November 11, 1983, with those results reported to Ohio University. The results are included in this report. The measured maximum lobe of radiation in the horizontal direction reported to Ohio University personnel on October 13, 1983, was 300 degrees, which is the azimuth used for the Clark tower and airborne measurements made by Ohio University on that date. For completeness of information the ISM device, Model B, did pass FCC testing performed by Elite on November 11, 1983. Tables 3 through 6 are the complete RF field intensities for the Clark tower measurements.

In spite of the foregoing exceptions, the quality and consistency of the testing indicate that these data do represent possible ranges of emission values obtainable from actual ISM equipment operation which was the goal of the study.
table 3. table of data for machine a clark tower measurements

| MACHINE MODEL: A OCT-12-1983 |  |  |  |  |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| dECAY EXPONENT $=1.95$ |  |  |  |  |  |  |
| RP | ELEV |  | $\mathrm{dB}_{\mu \mathrm{V}}$ | $\mathrm{dB}_{\mu} \mathrm{V}(1000)$ | $\mathrm{dB}_{\mu} \mathrm{V}$ | $\mathrm{dB}_{\mu} \mathrm{V}(1000)$ |
| SHIELDS | ANG | AZIMUTH | $\underline{27 \mathrm{MHz}}$ | 27 MHz | $\underline{109 \mathrm{NHz}}$ | $\underline{109 \mathrm{MHz}(1000)}$ |
| ON | 45. | 180. | 136.0 | 71.7 | 72.2 | 7.9 |
| ON | 59. | 180. | 126.0 | 66.8 | 73.3 | 14.1 |
| ON | 66. | 180. | 135.1 | 80.3 | 74.4 | 19.6 |
| ON | 71. | 180. | 134.4 | 83.2 | 73.3 | 22.1 |
| ON | 74. | 180. | 136.3 | 88.1 | 74.1 | 25.9 |
| Ofr | 66. | 180. | 144.6 | 89.8 | 75.3 | 20.5 |
| OfF | 74. | 180. | 146.4 | 98.2 | 71.8 | 23.6 |
| OSC OfF | 71. | 180. | 129.9 | 78.7 | 91.7 | 40.5 |
| OSC Off | 45. | 180. | 134.4 | 70.1 | 99.5 | 35.2 |
| OSC OFF | 59. | 180. | 114.4 | 55.2 | 94.2 | 35.0 |

table 4. table of data for machine b clark tower measurements

CLARK TOWER MEASUREMENTS
MACHINE MODEL: B
OCT-13-1983
dECAY EXPONENT = 1.57

| RF | Elev |  | $\mathrm{dB}_{\mu} \mathrm{V}$ | $\mathrm{dB} \mu \mathrm{V}(1000)$ | $\mathrm{dB}_{\mu} \mathrm{V}$ | $\mathrm{dB} \mu \mathrm{V}$ (1000) |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| SHIELDS | ANG | AZIMUTH | $\underline{27 \mathrm{MHz}}$ | 27 MHz | $\underline{109 \mathrm{MHz}}$ | $\underline{109 \mathrm{MHz}(1000)}$ |
| ON | 45. | 0. | 118.4 | 66.6 | 93.9 | 42.1 |
| ON | 59. | 0. | 105.0 | 57.4 | 101.0 | 53.4 |
| ON | 66. | 0. | 109.2 | 65.1 | 102.7 | 58.6 |
| ON | 71. | 0. | 107.8 | 66.6 | 103.6 | 62.4 |
| ON | 74. | 0. | 110.6 | 71.8 | 104.5 | 65.7 |
| ON | 74. | 300. | 109.2 | 70.4 | 103.6 | 64.8 |
| ON | 74. | 240. | 107.6 | 68.8 | 104.1 | 65.3 |
| ON | 71. | 240. | 102.3 | 61.1 | 103.5 | 62.3 |
| ON | 71. | 300. | 111.7 | 70.5 | 103.0 | 61.8 |
| ON | 66. | 300. | 105.7 | 61.6 | 98.9 | 54.8 |
| ON | 66. | 240. | 112.4 | 68.3 | 100.6 | 56.5 |
| ON | 59. | 240. | 105.6 | 58.0 | 99.8 | 52.2 |
| ON | 59. | 300. | 104.3 | 56.7 | 100.6 | 53.0 |
| ON | 45. | 300. | 113.2 | 61.4 | 96.5 | 44.7 |
| ON | 45. | 240. | 101.1 | 49.3 | 92.7 | 40.9 |

table 5. table of data for machine c clark tower measurements

CLARK TOWER MEASUREMENTS
MACHINE : $O O D E L: C$
OCT-13-1983

DECAY EXPONENT $=1.95$

| RF | ELEV ANG | AZIMUTH | $\begin{array}{r} \mathrm{dB} \mu \mathrm{~V} \\ 27 \mathrm{MHz} \mathrm{Z} \end{array}$ | $\begin{array}{r} \mathrm{dB} \mu \mathrm{~V}(1000) \\ 27 \mathrm{MHz} \end{array}$ | $\begin{array}{r} \mathrm{dB} \mathrm{\mu V} \\ 109 \mathrm{MHz} \end{array}$ | $\begin{array}{r} \mathrm{dB} \mu \mathrm{~V}(1000) \\ 109 \mathrm{MHz}(1000) \end{array}$ |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| OFF | 45. | 320. | 122.8 | 58.5 | 68.1 | 3.8 |
| OFF | 66. | 320. | 119.7 | 64.9 | 66.3 | 11.5 |
| OFF | 74. | 320. | 116.9 | 68.7 | 63.9 | 15.7 |
| ON | 74. | 260. | 100.5 | 52.3 | 57.4 | 9.2 |
| ON | 74. | 320. | 96.7 | 48.5 | 63.1 | 14.9 |
| ON | 74. | 20. | 96.4 | 48.2 | 66.5 | 18.3 |
| ON | 59. | 20. | 100.8 | 41.6 | 67.9 | 8.7 |
| ON | 59. | 320. | 88.3 | 29.1 | 63.5 | 4.3 |
| ON | 59. | 260. | 87.3 | 28.1 | 58.0 | -1.2 |
| ON | 45. | 260. | 100.9 | 36.6 | 64.9 | 0.6 |
| ON | 45. | 320. | 101.9 | 37.6 | 64.3 | -0.0 |
| ON | 45. | 20. | 109.4 | 45.1 | 65.5 | 1.2 |

TABLE 6. TABLE OF DATA FOR MACHINE D CLARK TOWER MEASUREMENTS

CLARK TOWER MEASUREMENTS
MACHINE MODEL: D
OCT-13-1983

DECAY EXPONENT = 1.95

| RF <br> SHIELDS | $\begin{aligned} & \text { ELEV } \\ & \text { ANG } \end{aligned}$ | AZIMUTH | $\begin{array}{r} \mathrm{dB} \mu \mathrm{~V} \\ 27 \mathrm{MHz} \end{array}$ | $\begin{array}{r} \mathrm{dB} \mu \mathrm{~V}(1000) \\ 27 \mathrm{MHz} \\ \hline \end{array}$ | $\begin{array}{r} \mathrm{dB} \mu \mathrm{~V} \\ 109 \mathrm{MHz} \end{array}$ | $\begin{array}{r} \mathrm{dB} \mu \mathrm{~V}(1000) \\ 109 \mathrm{MHz}(1000) \\ \hline \end{array}$ |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| OFF | 74. | 320. | 115.3 | 67.1 | 60.5 | 12.3 |
| OFF | 66. | 320. | 114.7 | 59.9 | 63.6 | 8.8 |
| OFF | 45. | 320. | 119.7 | 55.4 | 66.0 | 1.7 |
| ON | 45. | 20. | 105.0 | 40.7 | 64.4 | 0.1 |
| ON | 45. | 320. | 97.3 | 33.0 | 64.0 | -0.3 |
| ON | 45. | 260. | 92.9 | 28.6 | 61.5 | -2.8 |
| ON | 59. | 260. | 83.8 | 24.6 | 57.4 | -1.8 |
| ON | 59. | 320. | 89.5 | 30.3 | 62.1 | 2.9 |
| ON | 59. | 20. | 97.5 | 38.3 | 64.7 | 5.5 |
| ON | 74. | 20. | 99.1 | 50.9 | 64.5 | 16.3 |
| ON | 74. | 320. | 93.1 | 44.9 | 62.6 | 14.4 |
| ON | 74. | 260. | 99.3 | 51.1 | 56.9 | 8.7 |

VII. CO-CHANNEL INTERPERENCE

The co-channel interference effects to ILS localizers from nonaviation RF radlation sources have been addressed in recent work completed by the International Civil Aviation Organization (ICAO) [5,6,7]. Additional work completed regarding co-channel interference effects on VOR signals from CATV is also pertinent, and indicates very aimilar desired-to-undesired signal criteria to provide interference protection to localizers [8].

ICAO has defined four types of co-channel signals. Three of these types deal with unmodulated signals, and the remaining type involves modulated signals. In all cases, the specified desired/undesired ( $D / U$ ) signal levels indicate faterference that will cause no more than $5 \mu \mathrm{a}$ of localizer course deviation. The four types of interfering signals referred to by ICAO are summarized below.

## Unmodulated Carrier Interference:

TYPE I An unmodulated carrier within the localizer receiver RF passband and within 0.5 Hz of the 90 or 150 Hz sideband modulation of the ILS localizer must be as low as 46 dB below the desired localizer carrier.

TYPE II An unmodulated carrier within the localizer receiver RF passband and within 10 Hz of the 90 or 150 Hz sidebands, but not within the TYPE I tolerance, must be as low as 26 dB below the desired localizer carrier.

TYPE III An unmodulated carrier except TYPE I and TYPE II within the localizer receiver $R F$ passband with sufficient strength will cause progressive capturing of the receiver. The unwanted $R F$ signal field strength must be as low as 7 dB below the desired localizer carrier level.

## Modulated Carrier Interference:

TYPE IV An unwanted carrier except TYPE I and TYPE II with $20 \%$ amplitude modulation by either 90 or 150 Hz components must be as low as 13 dB below the desired localizer carrier level.

In general, any of the first three types are possible as interference to localizers from ISM equipment. The equipment tested during the contract produced only CW emissions.

To correlate the findings of this report to co-channel interference, a worst case example will be considered. The example is based on placing the ISM equipment at a point below a localizer approach course, located 3800 feet from the threshold of the runway and 200 feet below the glide path. A runway length of 7000 feet is assumed, with the ISM device producing co-
channel interference at the FCC or CISPR emissions limits. Table 7 indicates whether the ICAO D/U signal level criteris is satisfied. The condicions for the comparison are included in the cable. Measurements were made of the RP field strength of a localizer operating at the Ohio University Airport with a value of $91 \mathrm{~dB} \mu \mathrm{~V} / \mathrm{m}$ measured at the point in the approach indicated above. This signal level is sigaificantiy higher than the level considered as a minimum in ICAO Annex 10, Volume I, Part I, of 46 $\mathrm{dB} \mu \mathrm{V} / \mathrm{m}$ [9] at the threshold.

The results indicated in Table 7 represent the. ings from the open field measurements performed at Waterman, Illinois; the airborne measured RF fields are generally 20 to 40 dB above those measured on the ground. These results indicate that CISPR emissions limits do provide sufficient D/U levels except for the signal levels measured in the airborne tests considering TYPE I interference. For FCC enissions limits the results are quite different. All but one of the measured conditions fail the criteria for D/U signal levels. The one condition that did exceed the D/U level was for the TYPE III interference.

TAbLE 7. pass fail for icao interference desired-TO-UNDESIREd SIGNAL CRITERIA EXAMPLE


The authors would like to thank the following persons for their efforts: Dr. Robert Lilley for his skill as the pilot of the aircraft during the long hours of the flight tests in Waterman, Illinois; Mr. Richard Zoulek who completed the mechanical details of the flight instrumentation package and the preparation of the Clark tower; and Mr . James Klouda and Mr. John Modica for their work at the test site in Waterman, Illinois. We all worked long hours to complete the tests on time. Finally, I would like to thank Mr. Robert Smith of the FAA for his efforts in speeding up the necessary paper work required by the contract.

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X. APPENDIXES

## Appendix A.

This appendix contains all flight measurement plots for the open field testing of the four ISM devices tested at Waterman, Illinois. These are the plots of absolute field strength in $\mathrm{dB} \mu \mathrm{V} / \mathrm{m}$ vs. horizontal position of the aircraft over the ground. The plots have superimposed on them the RF radiation limits for FCC and CISPR for easy interpretation of the data relative to these limits. All distances are expressed in meters. These plots are referred to in the text of the report.


Figure A-1. Flight Data Machine A, $25 \mathrm{~kW}, 152 \mathrm{M}$ Altitude, $180^{\circ}$ Azimuth, RFI Shields Removed


Figure A-2. Flight Dato Machine A, $25 \mathrm{~kW}, 457 \mathrm{M}$ Altitude, $180^{\circ}$ Azimuth, RFI Shields Removed


Figure A-3. Flight Data Machine A, $25 \mathrm{~kW}, 152 \mathrm{M}$ Altitude, $180^{\circ}$ Azimuth, RFI Shields Removed


Figure A-4. Flight Data Machine A, $25 \mathrm{~kW}, 1.52 \mathrm{M}$ Altitude, $180^{\circ}$ Azimuth, RFI Shields Removed


Figure A-5. Flight Data Machine A, $25 \mathrm{~kW}, 152 \mathrm{M}$ Altitude, $180^{\circ}$ Azimuth, RFI Shields in Place

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$\square$


Figure A-6. Flight Data Machine B, $2 \mathrm{~kW}, 152 \mathrm{MAltitude}$, $300^{\circ}$ Azimuth, RFI Shields In Place


Figure A-7. Flight Data Machine B, $2 \mathrm{~kW}, 457 \mathrm{M}$ Altitude, $300^{\circ}$ Azimuth, RFI Shields In Place


Figure A-8. Flight Data Machine B, $2 \mathrm{~kW}, 457 \mathrm{M}$ Altitude, $0^{\circ}$ Azimuth, RFI Shields In Place


Figure A-9. Flight Data Machine B, $2 \mathrm{~kW}, 457 \mathrm{M}$ Altitude, $240^{\circ}$ Azimuth, RFI Shields In Place


Figure A-10. Flight Data Machine B, 2 kW , 152 M Altitude, $240^{\circ}$ Azimuth, RFI Shields In Place


Figure A-11. Flight Data Mochine B, $2 \mathrm{~kW}, 152 \mathrm{M}$ Altitude, $0^{\circ}$ Azimuth, RFI Shields In Place


Figure A-12. Flight Data Machine C, $3 \mathrm{~kW}, 152 \mathrm{M}$ Altitude, $320^{\circ}$ Azimuth, RFI Shields In Place


Figure A-13. Flight Dara Machine C, $3 \mathrm{~kW}, 152 \mathrm{MAltitude}$, $20^{\circ}$ Azimuth, RFI Shields in Place
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Figure A-14. Flight Data Machine C, $3 \mathrm{~kW}, 152 \mathrm{M}$ Altitude, $260^{\circ}$ Azimuth RFI Shields In Place


Figure A-15. Flight Data Machine C, $3 \mathrm{~kW}, 152 \mathrm{M}$ Altitude, $260^{\circ}$ Azimuth, RFI Shields Removed


Figure A-16. Flight Data Machine C, $3 \mathrm{~kW}, 152 \mathrm{M}$ Altitude, $320^{\circ}$ Azimuth, RFI Shields Removed


Figure A-17. Flight Data Machine C, $3 \mathrm{~kW}, 152 \mathrm{M}$ Altitude, $20^{\circ}$ Azimuth, RFI Shields Removed


Figure A-18. Flight Data Machine D, $2 \mathrm{~kW}, 152 \mathrm{M}$ Altitude, $320^{\circ}$ Azimuth, RFI Shields In Place


Figure A-19. Flight Data Machine D, $2 \mathrm{~kW}, 152 \mathrm{MAltitude}$, $20^{\circ}$ Azimuth, RFI Shields In Place


Figure A-20. Flight Data Machine D, $2 \mathrm{~kW}, 152 \mathrm{M}$ Altitude, $260^{\circ}$ Azimuth, RFI Shields In Place


Figure A-21. Flight Data Machine D, $2 \mathrm{~kW}, 152 \mathrm{M}$ Altitude, $200^{\circ}$ Azimuth, RFI Shields In Place


Figure A-22. Flight Doto Machine D, $2 \mathrm{~kW}, 152 \mathrm{M}$ Altitude, $60^{\circ}$ Azimuth, RFI Shields In Place

## Appendix B.

The material included here is taken from the reports furnished to Ohio University by the Elite Electronic Engineering Company. The test methods and results of the PCC, Part 18, tests conducted at their Waterman, Illinois, open field test site are described here. All of the measurement data for each of the ISM devices tested is included in this appendix except for the text describing the test procedure and equipment. This information is described in section IV of this report.

## ground rf field measurements - machine a



Figure B-1. Machine A Ground Determined Decay Exporvent


Figure B-2. Machine A Operating Frequency Field Intensity at 1000 feet


Figure B-3. Machine A Field Intensity vs. Frequency

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Test Distanr：： 200 ft．Azimuth：足 0 degrees Corrections hased an a field deray exponent of 1.95

| rreq． | Mtr RHg | Ant． | Dist． | Tntat | Total． | Limit |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
|  |  | far． | corr | OHuV／m | UV／M | UU／M |
| Mitr | H\＆いい | HF | UF | （1） 1 mila | ¢ 1mil？ | e lmile |


| 2．7． 2.390 | 73．9 | 11.0 | －－5\％． 4 | 28.5 | 26．：5 | 13.0 |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 54.4580 | 3P．5 | 13．0 | －55． 4 | $-10.0$ | 0.3 | 10.0 |
| 81．63370 | 3．） 1 | 8.7 | －55． 4 | －10．6 | 0.3 | 10.0 |
| 108.9160 | 40.8 | 11.9 | －55．4 | －7．8i | 0.7 | 10.0 |
| 13：1450 | $=5.6$ | 12.3 | －5s． 4 | 10.5 | 3.4 | 10.0 |
| 163.3739 | 54.0 | 19.5 | －55． 4 | 18：0 | 8.0 | 10.0 |
| 170．803．9 | ．54． 1 | 19.3 | $\cdots .53 .4$ | 1．4 | 1． $2 .$. | 10.0 |
| 21\％．8315 | 39.1 | 16.7 | －55．4 | 0.4 | 1.0 | 10.0 |
| 24：5， 06107 | ．33．7 | 17.0 | －55． 4 | －4． 7 | 0.6 | 10.0 |
| 372．2899 | 31.3 | 17.3 | －55． 4 | －6． 9 | 0.5 | 10.0 |

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| DATF TESTED | OCTOEER 11, 198\% |  |

「est Distance: 200 ft : Azimin: 40 degrees Corrections based on a field deray exponent of 1.95


| 227.25837 | 50.5 | 11.0 | -55. 4 | 6.1 | 2.0 | 0.0 |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 54.5174 | 30.1 | 12.9 | -55. 4 | - 92.4 | 0.2 | 10.0 |
| -1.7761 | 37.2 | 13.7 | $-5.5 .4$ | $-19.5$ | 0.1 | 10.0 |
| 109.0347 | 13.3 | 11.9 | -5F. 4 | $-31.2$ | 0.0 | 10.0 |
| 1.3大.?934 | 36.6 | 12.4 | -5.5. 4 | -6, 5 | 0.5 | 10.0 |
| 163.5521 | 30.2 | 19.5 | -55. 4 | - 5. 7 | 0.5 | 10.0 |
| 150.91108 | 25.9 | 183.3 | --5.5. 4 | -11.3 | 0.3 | 10.0 |
| 218.0695 | 43.4 | 16.7 | -55.4 | ?. 7 | 1.5 | 10.0 |
| 245.324\% | 34.10 | 17.0 | -5.5.4 | -14.4 | 0.8 | 10.0 |
| 272.5869 | 20.5 | 87.3 | -55.4 | $-17.7$ | 0.1 | 10.0 |

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rest Distance: 200 ft : Azimuth: Gll degreps Corrections based on a field deray exponent of 1.95



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| MOMM：＊ | A |  |
| S／N | PROTOTYPE |  |
| DATF TESTED | OCTOAFR 11，1983 |  |

Test Distance： $300 f$ f．Arimuth ：got degrees
Corrections based on a field decay exponent of 1.95

| Fran． | Mtr Racs | Ant． <br> fac | Dirt. <br> corr | Total． druv／ri | rotal． <br> uU／i | Liblat <br> UU／M |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| Mils | H3， | Ha | 18 | M IMillo | © 1miln | e lmile |


| 27．． 3.377 | 71.1 | 11.0 | －5．5． 4 | 2．6． 7 | 21.5 | 0.0 |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 54.4753 | 19.7 | 12.9 | －55．4 | －23．8 | 0.1 | 10.0 |
| 81.71 .30 | 40.6 | H． 7 | －－゙大 4 | $-6.1$ | 0.5 | 10.0 |
| 108.9505 | 36.9 | 11.9 | －55．4 | －6．7 | 0.5 | 10.0 |
| 1．35． 13383 | 54．ts | 13.4 | $\cdots 5.4$ | 11.5 | 3.8 | 10.0 |
| 163．4259 | 47.9 | 19.5 | －365． 4 | 31.9 | 4.0 | 10.0 |
| 190．65．36 | 3．3． 1 | 18．8？ | $-5.5 .4$ | －5． 1 | 0.6 | 10.0 |
| 217.9013 | 39.2 | 16.7 | －5ict 4 | 0.5 | 1.1 | 10.0 |
| ？ 35.1384 | 2：3．83 | 17.0 | －55． 4 | －15．6 | 0.2 | 10.0 |
| ア7コ．3766 | 33.9 | 17.3 | －55． 4 | －4．3 | 0.6 | 10.0 |




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| 27.1 .314 | 5i） 3 | 11.0 | －55．4 | 12． 4 | 4.1 | 0.0 |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 54.2627 | $3 \% .1$ | 13.0 | －55． 4 | －5．3 | 0.5 | 10.0 |
| 81.3741 | 33． 9 | 3.7 | －5．5．4 | $-13.9$ | 0.2 | 10.0 |
| $108: .5254$ | 13.9 | 11．8 | －55． 4 | －－99．7 | 0.0 | 10.0 |
| 1.35 .656 .3 | 3．3．7 | 13.8 | －5\％， 4 | －9．5 | 0.3 | 10.0 |
| 168．788\％ | \％9．6 | 19.3 | －55． 4 | －6． | 0.8 | 10.0 |
| 18\％．719\％ | ；3．？ | 1：3．4 | －5以 ． 4 | －7．9 | 0.4 | 10.0 |
| 21\％．0509 | 0.8 .9 | 16.7 | －5\％， 4 | －9．8 | 0.3 | 10.0 |
| ？ 44.1323 | 17.3 | 17.1 | －5．5． 4 | $-19.1$ | 0.1 | 10.0 |
| 271.3136 | 19.8 | 17.3 | －55． 4 | $-18.4$ | 0.1 | 10.0 |

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:SN : PROTOTYPE
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fent Distance: 200 ft : Azinuth : 141] degrees Corrections based on a field decay exponent of $1.9 \mathrm{E}_{\mathrm{i}}$

| Fra | Mtr RHg | Ant. <br> fac. | Dist. <br> corr | Total dRuU/m | Total HV/M | Limit UV/M |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| ifty: | nsal | 18 | 116 | * Imila | ம¢ IMila | e Imil |


| 27.1.39:3 | 56.4 | 11.0 | -55. 4 | 12.0 | 4.1) | 12.0 |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 54.2795 | 40.8 | 13.0 | -55.4 | $-1.6$ | 0.8 | 10.0 |
| 01.419 .3 | 37.7 | 3.7 | -55. 4 | -3.7 | 0.4 | 10.0 |
| 109.5590 | 30.1 | 11.8 | -55.4 | -73.E | 0.1 | 10.0 |
| 1.35. 6.733 | 3-5.5 | 12.2 | --5\% , 4 | -7.7 | 0.4 | 10.0 |
| 16 ¢.. 6335 | 2\%.\% | 19.3 | -55.4 | -8.4 | 0.4 | 10.3 |
| 139.3733 | 2\%.0 | 13.3 | -5\%. 4 | -12.1 |  | 10.0 |
| 21\%.1180 | 25.5 | 16.7 | -55.4 | $-13.2$ | 0.2 | 10.0 |
| 2.44.357:3 | 16, 5 | 17.0 | -55.4 | $-2.1 .9$ | 0.1 | 10.0 |
| 271.39\%5 | 20.4 | 17.3 | -55.4 | -17.8 | 0.1 | 10.0 |

 D\&TA : AM,

 MTINFL $\ddagger: A$ $S / N$ DATF TCETFD : OCTOAER 11, 1983

Test Diriance: 200 ft. Azifuth : iAD degraes Corrections hased on field decay exponent of 1.55

| Frert. | Mtr ROg | Ant. | Dist. | Total | rotal. | Limit |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
|  |  | far. | corr | dBuV/m | uU/m | uV/m |
| MHy | aroue | H3 | 18 | P 1 milip | ¢¢ 1mila | E lmile |


| 27.1197 | 57, 2 | 11.0 | -55. 4 | 12.8 | 4.3 | 0.0 |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 54.2394 | 43.1 | 13.0 | -53.4 | 0.7 | 1.1 | 10.0 |
| 81.35178 | 41.3 | 8.7 | -5.5. 4 | -6.0 | 0.5 | 10.0 |
| 101:,4787 | 8.9 | 11.8 | -55. 4 | -80.7 | 0.1 | 10.0 |
| 1.35: , 57834 | ? $3: 3.7$ | 12. ${ }^{\text {a }}$ ? | -5.5. 4 | -14.4 | 0.8 | 10.0 |
| 168... 7181 | 2 Cl 7 | 17.3 | -55.4 | $-13.4$ | 0.8 | 10.0 |
| 18\%.8.37\% | 2A, 1 | 1:3.4 | -55. 4 | $-11.1$ | 0.13 | 10.0 |
| 216.9574 | -6.1 | 16.7 | -55. 4 | $-12.6$ | 0.2 | 10.0 |
| \$344.07\%1 | 17.9 | 17.0 | -5.5. 4 | -20.5 | 0.1 | 10.0 |
| ¢71.1568 | 17.2 | 17.3 | -SE. 4 | -2.1.0 | 0.1 | 10.0 |

ET: 8198

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Ti.:. 1

MANATHA: H15F:ه M(i]) $: ~=A$
$: S / N$ : PROTOTYPE
DATF TESTKD : OCTONKR 11, 19\&i3

> Iest Distanre : 200 ft. Arimuth : linl degraes Correctinns hased on a field decay exponent of 1.8 F



ETE 8198



TI: T
MANIIFAI: THRI:R MIIDEL * $: 3 / N$ PROTOTYPE DATE TESTED
 DATE TEGTED OCTTAEFR 11, 190.

Test Distance: 200 ft. Arimuth : 200 degrepes Corrections based on a fiejd deray expanent of 1.95




MANIIHARTIJK:C: MTいITL : A S/N

PROTQTYPE
OCTORFR 11, 198:3

DATF. TESTFD

A
rest Distance: 200 ft. Azitutin : 2fill degreps Corrections based on a fiejd decay exponent of 1.95



$$
\begin{aligned}
& \text { 1.in } 8198
\end{aligned}
$$


MFinllf:AC,T1Hiz:..R MCIDFL : A SS/N A
PROTOTYPE
DATF TESTEA : OCTOTER 13, 19E:3
rest Distance: $200 f t$ Azimuth : 24n degreas Corrections based on a field decas exporient of 1.7 in $^{\prime}$

27.1401
54.2801
81.42002
104.5603
1.35 .7004
168.8404
187.9805
217.1206
3.44 .2607
271.4007
6.3 .9
39.8
37.7
30.3
313.5
29.9
26.3
31.6
33.2 .1
17.7

| 11.1 | -55.4 |
| ---: | ---: |
| 13.0 | -55.4 |
| 3.7 | -55.4 |
| 11.8 | -5.4 .4 |
| 13.3 | -5.5 .4 |
| 19.3 | -55.4 |
| 18.3 | -55.4 |
| 16.7 | -5.5 .4 |
| 17.0 | -65.4 |
| 17.3 | -55.4 |

17.5
-3.2
-7.8
-33.3
-4.7
-6.2
-10.83
-7.1
-14.3
-20.5

| 9.4 | 0.0 |
| ---: | ---: |
| 0.7 | 10.0 |
| 0.4 | 10.0 |
| 0.1 | 10.0 |
| 0.6 | 10.0 |
| 0.5 | 10.0 |
| 0.3 | 10.0 |
| 0.4 | 10.0 |
| 0.8 | 10.0 |
| 0.1 | 10.0 |

## ET. 8198

 DATA :-AB:


Test Distanre: 200 ft. Arimuth : 260 degrees Corrections hased on a fipld decay exponent of $1.9 \%$

| Freq. | Mtr Rdg | Ant. <br> fac. | $\begin{aligned} & \text { Dist. } \\ & \text { corr } \end{aligned}$ | Total dFuV/M | Total UV/M | Limit uU/M |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| Mits. | AFul | 1F | dis | - imile | ¢ 1mila | e lmile |


| 27.2421 | 77.3 | 11.0 | -5.5. 4 | 32.9 | 43.9 | 0.0 |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 54.4842 | 32.2 | 13.9 | -5.5.4 | -10.3 | 0.3 | 10.0 |
| 81. 72.7? | 36. 9 | 8.7 | -5\%. 4 | -9.8 | 0.3 | 10.0 |
| 108:.9683 | 24.8 | 11.9 | -5.5.4 | $\therefore 18.8$ | 0.1 | 10.0 |
| 136.3104 | $4 \% .0$ | 13.4 | -.5.5.4 | 3.8 | 1. 1. | 10.0 |
| 163.4525 | 50.8 | 19.5 | -55. 4 | 14.8 | 5.5 | 10.0 |
| 191.694t | 23. 1 | 18.? | -5.5. 4 | $-12.1$ | 0.2 ? | 10.0 |
| 21\%.9366 | 38.3 | 16.7 | -55.4 | -0.4 | 1.0 | 10.0 |
| 245.1787 | 3.7.7 | 17.0 | -5\%i. 4 | -4.7 | 0.6 | 10.0 |
| 272.4208 | 34.8 | 17.3 | -55. 4 | -3.4 | 0.7 | 10.0 |

## ETP 8198

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MANHFMCTIRR: is MOIH:L : : $: / \mathrm{N}$ DATE TESTFD


## A

PROTGTYPE
OCTHRHR 11, 1983

Test Distance: 200 ft. Arimuth : 2i30 degrees Corrections hased on a fipld decay exponent of 1.95

| Freq. | Mtr Rdij | $\begin{aligned} & \text { Ant. } \\ & \text { fac. } \end{aligned}$ | $\begin{aligned} & \text { Diat. } \\ & \text { corr } \end{aligned}$ | Total aBuV/M | Total UU/M | Limit UV/m |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| Mily | divu | Ha | HE | - imile | p imile | e imile |


| 27.1583 | 67.2 | 11.0 | .-55. 4 | 22.8 | 13.7 | 13.0 |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 54.3046 | 31.9 | 13.0 | -55.4 | -10.5 | 0.3 | 10.0 |
| 81.4549 | 34.5 | 3.7 | -55.4 | -12.3 | 0.2 | 10.0 |
| 108.6092 | 19.0 | 11.5 | -5.5.4 | -34.6 | 0.1 | 10.0 |
| 135.7615 | 4.3 .2 | 12, 3 | - ¢\% . 4 | -0.0 | 1.0 | 10.0 |
| 16 ¢. 9138 | 24.3 | 19.4 | -55.4 | -11.8 | 0.3 | 10.0 |
| 170.0661 | 29.7 | 18.3 | -3:3.4 | -7.4 | 0.4 | 10.0 |
| 217.2184 | 77.0 | 16.7 | -55. 4 | -11.7 | 0.3 | 10.0 |
| 244.3707 | 19.1 | 17.0 | --5\% . 4 | $-19.0$ | 0.1 | 10.0 |
| 271.5230 | 19.8 | 17.3 | -55.4 | $-18.4$ | 0.1 | 10.0 |

ET: 8198

rifici: :Ans.


> Test Distanee: 200 ft. Aximuth : 300 degrees Corrections hased on a field decay exponent of 1.95


| 27.245 .3 | 75.7 | 11.0 | -55.4 | 32.3 | 41.0 | 0.0 |
| ---: | ---: | ---: | ---: | ---: | ---: | ---: |
| 54.4905 | 39.8 | 12.9 | -55.4 | -2.7 | 0.7 | 10.0 |
| 81.7353 | 4.3 .7 | 3.7 | -55.4 | -3.0 | 0.7 | 10.0 |
| $104: .9811$ | 30.1 | 11.9 | -55.4 | -13.5 | 0.2 | 10.0 |
| 134.3263 | 5.3 .0 | 12.4 | -5.5 .4 | 9.9 | 3.1 | 10.0 |
| 163.4716 | 55.3 | 19.5 | -55.4 | 17.3 | 9.3 | 10.0 |
| 170.7169 | 34.2 | 18.2 | -5.5 .4 | -3.0 | 0.7 | 10.0 |
| 217.9522 | 52.2 | 16.7 | -55.4 | 13.5 | 4.7 | 10.0 |
| 245.3074 | 33.13 | 17.0 | -55.4 | 0.4 | 1.0 | 10.0 |
| 272.4527 | 44.3 | 17.3 | -55.4 | 6.1 | 2.0 | 10.0 |

## ETH 8198

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MANHRAISTIJR: O : MOXICL : A $S / N$ PROTOTYPE DATE TESTED : OCTOFER 11, 19KZ

```
Test Distanre : 200 ft. Azimuth ; 3:00 degrees
Correctjons based on a field decay exponent of 1.95
```

| Frs\%. | Mtr RHg | Ant. | Dist. | Tntal. | Topal. | Limit |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
|  |  | fac. | corr | dFuV/m | UV/m | UU/M |
| MHT. | dBul | ds | dS | ¢ 1 inile |  | e lmile |


| 27.2502 | 75.7 | 11.0 | -55.4 | 31.3 | 36.5 | 0.0 |
| ---: | ---: | ---: | ---: | ---: | ---: | ---: |
| 54.5004 | 38.8 | 12.9 | -55.4 | -3.7 | 0.7 | 10.0 |
| 181.7507 | 45.5 | 8.7 | -55.4 | -1.2 | 0.7 | 10.0 |
| 105.0009 | 34.0 | 11.9 | -55.4 | -9.5 | 0.3 | 10.0 |
| 136.3 .511 | 55.7 | 12.4 | -55.4 | 12.6 | 4.3 | 10.0 |
| 163.5013 | 50.8 | 19.5 | -55.4 | 14.9 | 5.5 | 10.0 |
| 190.7515 | 40.9 | 17.2 | -5.5 .4 | 3.7 | 1.5 | 10.0 |
| 215.0017 | 54.8 | 16.7 | -55.4 | 16.1 | 6.4 | 10.0 |
| 245.2520 | 37.1 | 17.0 | -55.4 | -1.3 | 0.7 | 10.0 |
| 272.5024 | 40.0 | 17.3 | -55.4 | 1.8 | 1.2 | 10.0 |

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Test Distance: 200 fr. Azimuth : $34 n$ degrees
Corrections hased on a field decay exponent of 1.95


| 27.1571 | 64.1 | 11.0 | -55.4 | 19.7 | 9.8 | 0.0 |
| ---: | ---: | ---: | ---: | ---: | ---: | ---: |
| 54.3143 | 37.5 | 13.0 | -55.4 | -4.9 | 0.6 | 10.0 |
| 81.4714 | 44.7 | 8.7 | -53.4 | -2.1 | 0.13 | 10.0 |
| 108.6286 | 2.9 .1 | 11.9 | -55.4 | -14.5 | 0.8 | 10.0 |
| 135.7857 | 37.5 | 12.2 | -55.4 | -5.6 | 0.5 | 10.0 |
| 16.9429 | 23.2 | 19.4 | -55.4 | -12.9 | 0.2 | 10.0 |
| 190.1000 | 24.6 | 18.3 | -55.4 | -12.5 | 0.2 | 10.0 |
| 217.2572 | 27.8 | 16.7 | -55.4 | -10.9 | 0.3 | 10.0 |
| 244.4143 | 20.3 | 17.0 | -55.4 | -17.6 | 0.1 | 10.0 |
| 271.5714 | 16.1 | 17.3 | -55.4 | -22.1 | 0.1 | 10.0 |

ground rf field measurements - machine b


Figure B-4. Machine B Ground Determined Decay Exponent

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Figure B-5. Machine B Operating Frequency Field Intensity of 1000 foet


Figure B-6. Machine B Field Intensity vs. Frequency

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EL IMF ELIFITRONIC. TNTINFIFKING CO. DATA "ACHE


Test Distance: 75 ft. Azimuth: 0 degrees Corrections based on field decay exponent of 1.95


| 27.0711 | 45.3 | 11.0 | -72.1 | -15.8 | 0.2 | 0.0 |
| ---: | ---: | ---: | ---: | ---: | ---: | ---: |
| 54.1821 | 31.3 | 13.1 | -72.1 | -27.7 | 0.0 | 10.0 |
| 81.2732 | 41.3 | 8.6 | -72.1 | -22.1 | 0.1 | 10.0 |
| 108.3642 | 71.3 | 11.8 | -72.1 | 11.1 | 3.6 | 10.0 |
| 135.4553 | 54.3 | 12.1 | -72.1 | -.5 .1 | 0.6 | 10.0 |
| 162.5463 | 55.9 | 19.3 | -72.1 | 3.1 | 1.4 | 10.0 |
| 189.6374 | 48.9 | 18.4 | -72.1 | -4.8 | 0.6 | 10.0 |
| 216.7284 | 42.7 | 16.7 | -72.1 | -12.6 | 0.2 | 10.0 |
| 243.8195 | 40.0 | 17.0 | -72.1 | -15.1 | 0.2 | 10.0 |
| 270.9105 | 39.0 | 17.3 | -72.1 | -15.8 | 0.2 | 10.0 |

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Test Distance: 75 ft. Azimuth : 20 degreas Corrections based on a field decay exponent of 1.95


| 27.1457 | 53.0 | 11.0 | -72.1 | -13.1 | 0.1 | 0.0 |
| ---: | ---: | ---: | ---: | ---: | ---: | ---: |
| 54.2915 | 23.9 | 13.0 | -72.1 | -35.1 | 0.0 | 10.0 |
| 61.4372 | 38.2 | 8.7 | -72.1 | -25.2 | 0.1 | 10.0 |
| 101.5829 | 70.6 | 11.5 | -72.1 | 10.4 | 3.2 | 10.0 |
| 135.73 .87 | 53.1 | 12.2 | -72.1 | -6.7 | 0.5 | 10.0 |
| 16.5 .8744 | 5.5 .4 | 19.3 | -72.1 | 2.7 | 1.4 | 10.0 |
| 190.0301 | 51.1 | 18.3 | -72.1 | -2.6 | 0.7 | 10.0 |
| 217.1659 | 42.7 | 16.7 | -72.1 | -12.6 | 0.8 | 10.0 |
| 244.3116 | 41.8 | 17.0 | -72.1 | -13.3 | 0.2 | 10.0 |
| 271.4573 | 37.5 | 17.3 | -72.1 | -17.3 | 0.1 | 10.0 |

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DATE TESTED
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PROTOTYRE NOUEMFER 11, 1983

Test Distance: 75 ft. Azimuth: 40 degraes Coreections based on field decay exponent of 1.95

| Freq. | Mtr Rdg | Ant. fac | Dist. <br> rorr | Tntal dRuV/m | Total $\mathrm{l} \mathrm{V} / \mathrm{m}$ | $\begin{aligned} & \text { Limit } \\ & \text { UU/M } \end{aligned}$ |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| MHz | AEUV | ds | d8 | E Imile | e 1milp | e 1mile |


| 27.1234 | 55.1 | 11.0 | -72.1 | -6.0 | 0.5 | 0.0 |
| ---: | ---: | ---: | ---: | ---: | ---: | ---: |
| 54.2468 | 26.3 | 13.0 | -72.1 | -32.7 | 0.0 | 10.0 |
| 81.3702 | 38.4 | 8.7 | -72.1 | -25.0 | 0.1 | 10.0 |
| 108.4936 | 68.6 | 11.8 | -72.1 | 8.4 | 2.6 | 10.0 |
| 13.6170 | 53.0 | 12.2 | -72.1 | -6.9 | 0.5 | 10.0 |
| 162.7404 | 5.4 .6 | 19.3 | -72.1 | 1.9 | 1.2 | 10.0 |
| 189.8438 | 52.8 | 18.4 | -72.1 | -0.9 | 0.9 | 10.0 |
| 216.9871 | 44.4 | 16.7 | -72.1 | -10.9 | 0.3 | 10.0 |
| 244.1105 | 43.6 | 17.0 | -72.1 | -11.5 | 0.3 | 10.0 |
| 271.2339 | 34.0 | 17.3 | -72.1 | -20.8 | 0.1 | 10.0 |

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ELITT EIT:TRONTS: RNGTNEERING CO. DAlA PARC

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TEST: FCC. FART ISN INDISTRIAL HIATJNG: FOIIMPMFNI
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MANIJFAR:TIRRFR: MODFL : B S/N : PROTOTYPE DATE TESTED : NOUEMEER 11, 1983

Test Distance: 7S ft. Azimuth: 60 degrees Corrections based on a firld decay exponent of 1.95


| 27.1120 | 54.0 | 11.0 | -72.1 | -7.1 | 0.4 | 0.0 |
| ---: | ---: | ---: | ---: | ---: | ---: | ---: |
| 54.2240 | 29.8 | 13.1 | -72.1 | -29.2 | 0.0 | 10.0 |
| 01.3360 | 39.5 | 8.7 | -72.1 | -23.9 | 0.1 | 10.0 |
| 108.4480 | 65.6 | 11.8 | -72.1 | 9.4 | 2.9 | 10.0 |
| 135.5579 | 50.2 | 12.2 | -72.1 | -9.7 | 0.3 | 10.0 |
| 162.6719 | 48.6 | 19.3 | -72.1 | -4.2 | 0.6 | 10.0 |
| 189.7337 | 40.0 | 48.4 | -72.1 | -13.7 | 0.2 | 10.0 |
| 216.8959 | 39.1 | 16.7 | -72.1 | -16.2 | 0.2 | 10.0 |
| 244.0079 | 46.2 | 17.0 | -72.1 | -8.7 | 0.4 | 10.0 |
| 271.1199 | 28.7 | 17.3 | -72.1 | -26.1 | 0.0 | 10.0 |

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Test Distance： 75 ft ：Azimuth ： 80 degrees Carrections based on field decay exponent of 1.95


| 27.1079 | 55.3 | 11.0 | -72.1 | -5.8 | 0.5 | 0.0 |
| ---: | ---: | ---: | ---: | ---: | ---: | ---: |
| 54.2158 | 31.2 | 13.1 | -72.1 | -27.8 | 0.0 | 10.0 |
| 81.3236 | 40.5 | 8.7 | -72.1 | -22.9 | 0.1 | 10.0 |
| 108.4315 | 67.7 | 11.8 | -72.1 | 7.5 | 2.4 | 10.0 |
| 135.5 .394 | 50.4 | 12.2 | -72.1 | -9.5 | 0.3 | 10.0 |
| 182.6473 | 47.1 | 19.3 | -72.1 | -5.7 | 0.5 | 10.0 |
| 187.755 .2 | 44.2 | 18.4 | -72.1 | -9.5 | 0.3 | 10.0 |
| 216.8630 | 40.7 | 16.7 | -72.1 | -14.6 | 0.2 | 10.0 |
| 24.3 .7707 | 44.0 | 17.0 | -72.1 | -11.1 | 0.3 | 10.0 |
| 271.0788 | 42.2 | 17.3 | -72.1 | -12.6 | 0.2 | 10.0 |



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Test Distance: 75 ft : Azimuth 120 degrees Corrections based on field decay exponent of 1.95


| 27.1120 | 51.4 | 11.0 | -72.1 | -9.7 | 0.3 | 0.0 |
| ---: | ---: | ---: | ---: | ---: | ---: | ---: |
| 54.2240 | 33.4 | 13.1 | -72.1 | -25.6 | 0.1 | 10.0 |
| 81.3360 | 30.3 | 9.7 | -72.1 | -33.1 | 0.0 | 10.0 |
| 108.4480 | 62.3 | 11.8 | -72.1 | 2.1 | 1.3 | 10.0 |
| 1.35 .5600 | 47.8 | 17.2 | -72.1 | -10.1 | 0.3 | 10.0 |
| 16.2 .6721 | 51.1 | 19.3. | -72.1 | -1.7 | 0.8 | 10.0 |
| 187.7841 | 40.7 | 13.4 | -72.1 | -13.0 | 0.2 | 10.0 |
| 216.8961 | 46.6 | 16.7 | -72.1 | -8.7 | 0.4 | 10.0 |
| 244.0081 | 35.5 | 17.0 | -72.1 | -19.6 | 0.1 | 10.0 |
| 271.1201 | 43.8 | 17.3 | -72.1 | -11.0 | 0.3 | 10.0 |

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Test Distance: 75 ft Azimuth i 141 degrees Corrections based on field decay exponent of 1.95


| 27.1111 | 50.4 | 11.0 | -72.1 | -10.7 | 0.3 | 0.0 |
| ---: | ---: | ---: | ---: | ---: | ---: | ---: | ---: |
| 54.2223 | 33.7 | 13.1 | -72.1 | -25.3 | 0.1 | 10.0 |
| 81.3334 | 27.4 | 8.7 | -72.1 | -34.0 | 0.0 | 10.0 |
| 106.4445 | 65.0 | 11.8 | -72.1 | 4.8 | 1.7 | 10.0 |
| 135.5556 | 49.1 | 12.2 | -72.1 | -10.8 | 0.3 | 10.0 |
| 162.6668 | 50.3 | 19.3 | -72.1 | -2.5 | 0.8 | 10.0 |
| 187.7777 | 40.2 | 18.4 | -72.1 | -13.5 | 0.2 | 10.0 |
| 216.8890 | 43.4 | 16.7 | -72.1 | -11.9 | 0.3 | 10.0 |
| 244.0002 | 34.1 | 17.0 | -72.1 | -.21 .0 | 0.1 | 10.0 |
| 271.1113 | 20.9 | 17.3 | -72.1 | -33.9 | 0.0 | 10.0 |



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DATE TESTED

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PROTOTYPE NOUEMEER 11，1983

Test Distance： 75 ft．AziAuth ith degrues Corrections based on field decay exponent of 1.95


| 27.1074 | 47.5 | 11.0 | -72.1 | $\therefore-13.6$ | 0.2 | 0.0 |
| ---: | ---: | ---: | ---: | ---: | ---: | ---: | ---: |
| 54.2148 | 32.7 | 13.1 | -72.1 | -26.3 | 0.0 | 10.0 |
| 81.32 .22 | 29.9 | 8.7 | -72.1 | -33.5 | 0.0 | 10.0 |
| 108.4295 | 65.7 | 11.8 | -72.1 | 5.5 | 1.9 | 10.0 |
| 135.5369 | 47.6 | 12.2 | -72.1 | -10.3 | 0.3 | 10.0 |
| 162.6443 | 48.4 | 15.3 | -72.1 | -4.4 | 0.6 | 10.0 |
| 189.7517 | 31.7 | 18.4 | -72.1 | -31.8 | 0.1 | 10.0 |
| 216.8591 | 42.9 | 16.7 | -72.1 | -12.4 | 0.2 | 10.0 |
| 243.9665 | 37.8 | 17.0 | -72.1 | -17.5 | 0.1 | 10.0 |
| 271.0739 | 39.6 | 17.3 | -72.1 | -15.2 | 0.2 | 10.0 |

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Test Distance: 75 ft. Azimuth : 180 degrees Corrections based on a field decay exponent of 1.95


| 27.1127 | 47.8 | 11.0 | -72.1 | -13.3 | 0.7 | 0.0 |  |
| ---: | ---: | ---: | ---: | ---: | ---: | ---: | ---: |
| 54.2254 | 28.6 | 13.1 | -72.1 | -2 | -30.4 | 0.0 | 10.0 |
| 31.3380 | 35.0 | 8.7 | -72.1 | -28.4 | 0.0 | 10.0 |  |
| 108.4507 | 65.8 | 11.8 | -72.1 | 5.6 | 1.9 | 10.0 |  |
| 135.5634 | 48.6 | 12.2 | -72.1 | -11.3 | 0.3 | 10.0 |  |
| 162.6761 | 47.2 | 19.3 | -72.1 | -5.6 | 0.5 | 10.0 |  |
| 139.7818 | 34.4 | 18.4 | -72.1 | -19.3 | 0.1 | 10.0 |  |
| 216.9014 | 30.2 | 16.7 | -72.1 | -25.1 | 0.1 | 10.0 |  |
| 244.0141 | 42.6 | 17.0 | -72.1 | -12.5 | 0.2 | 10.0 |  |
| 271.1268 | 42.4 | 17.3 | -72.1 | -12.4 | 0.2 | 10.0 |  |

「TK 日: (I)
 DATA PACC

TFGT : FCC PART 1RD INDIISTRTAI HFATING FGUITMINI MANIIFACTTURER: MODEL : B :S/N : PROTOTYPE DATE TESTED : NDVEMEER 11, 1983

Test Distance: 75 ft. Azimuth : 200 degrees Corrections based on a field decay exponent of 1.9 F

27.1214
54.2427
81. 3641
108.4854
135.6068
162.7282
189.8495
216.9709
244.0922
271.2136
51.3
21. 0
41.1
67.1
50.4
43.8
51.2
37.4 $46.7 \quad 17.0$
39.6
11.0
13.0
8.7
11.8
12.2
19.3
18.4
16.7
17.0
17.3
$-72.1$
$-9.8$
$-38.0$
$-22.3$
6.9
2.210 .0
$\begin{array}{lll}-8.9 & 0.4 & 10.0\end{array}$
$-2.5 \quad 0.8 \quad 10.0$
$-17.9 \quad 0.1 \quad 10.0$
$\begin{array}{rrr}-13.4 & 0.4 & 10.0 \\ -15.2 & 0.2 & 10.0\end{array}$
$-15.2$

| 0.3 | 0.0 |
| ---: | ---: |
| 0.0 | 10.0 |
| 0.1 | 10.0 |
| 2.2 | 10.0 |
| 0.3 | 10.0 |
| 0.4 | 10.0 |
| 0.8 | 10.0 |
| 0.1 | 10.0 |
| 0.4 | 10.0 |
| 0.2 | 10.0 |

rTr annj
RI ITE EII CTETINT: PNISNFFRING CEI.
DATA PAC:T

TFST
MANIIFAK.TURF:R MOIDEL * S/N DATE TESTED

FCR: PART 1 RD JNDISTRTAI IIFATINF FOUTHMFNT
8
PROTOTYPE NOUEMAER 11, 1983

Test Digtance $\quad 75 \mathrm{ft}$. Azimuin: 220 D degrees Corrections based on field decay exponent of 1.95


| 27.11 .37 | 52.2 | 11.0 | -72.1 | -8.7 | 0.4 | 0.0 |
| ---: | ---: | ---: | ---: | ---: | ---: | ---: |
| 54.2077 | 21.2 | 13.1 | -72.1 | -37.8 | 0.0 | 10.0 |
| 81.3116 | 43.0 | 8.7 | -72.1 | -20.4 | 0.1 | 10.0 |
| 108.4154 | 66.0 | 11.8 | -72.1 | 5.8 | 1.9 | 10.0 |
| 135.5193 | 45.9 | 12.2 | -72.1 | -14.0 | 0.8 | 10.0 |
| 162.6232 | 42.8 | 19.3 | -72.1 | -10.0 | 0.3 | 10.0 |
| 187.7270 | 44.3 | 18.4 | -72.1 | -7.4 | 0.3 | 10.0 |
| 216.8305 | 37.6 | 16.7 | -72.1 | -17.7 | 0.1 | 10.0 |
| 243.9347 | 43.1 | 17.0 | -72.1 | -18.0 | 0.3 | 10.0 |
| 271.0386 | 33.3 | 17.3 | -72.1 | -21.5 | 0.1 | 10.0 |

「"TR R?O!
FI ITE ELIF CTRONTE ENGTN: FITNG RG. DAIA PAGEL


Test Distance: 75 ft. Arimuth : 240 degrees Corrections based on a field decay exponent of 1.95


| 27.10 .35 | 54.2 | 11.0 | -72.1 | -6.9 | 0.5 | 0.0 |
| ---: | ---: | ---: | ---: | ---: | ---: | ---: |
| 54.2069 | 25.3 | 13.1 | -72.1 | -33.7 | 0.0 | 10.0 |
| 81.3104 | 39.7 | 8.7 | -72.1 | -23.5 | 0.1 | 10.0 |
| 108.4138 | 66.1 | 11.8 | -72.1 | 5.5 | 2.0 | 10.0 |
| 13.5 .5173 | 47.2 | 12.2 | -72.1 | -12.7 | 0.2 | 10.0 |
| 162.6208 | 45.1 | 19.3 | -72.1 | -7.7 | 0.4 | 10.0 |
| 189.7242 | 43.3 | 18.4 | -72.1 | -10.4 | 0.3 | 10.0 |
| 216.8277 | 47.3 | 16.7 | -72.1 | -8.0 | 0.4 | 10.0 |
| 243.9311 | 46.6 | 17.0 | -72.1 | -8.5 | 0.4 | 10.0 |
| 271.0346 | 29.1 | 17.3 | -72.1 | -25.7 | 0.1 | 10.0 |

「TV: Rent
LI ITH ELIFTPRNTE FNGTNTFISNG EU. DATA PAGE

TES:T : FC:C PART $18 D$ INDISGETAI HFATTNR FGUJFMIM MAHIHFAR.IIRER MODEL : B
G/N : PROTOTYPE DATE TESTED : NOUEMFER 11, 1983
Test Distance: 75 ft Azimuth 260 degrees
Corrections based on field decay exponent of 1.95
Freq. Mtr Rog Ant. Dist, Total Total

MHz $\quad d B a v$

$$
\begin{array}{cc}
\text { fac } & \text { Cner } \\
\text { HB }
\end{array}
$$

$$
\begin{array}{lll}
\text { Total } & \text { Total } & \text { Limit } \\
\text { dRuU/M } & \text { UV/m } & \text { UV/M }
\end{array}
$$

e lmile elmile olmile

| 27.1078 | 52.7 | 11.0 | -72.1 | -8.2 | 0.4 | 0.0 |
| ---: | ---: | ---: | ---: | ---: | ---: | ---: |
| 54.2155 | 27.0 | 13.1 | -72.1 | -32.0 | 0.0 | 10.0 |
| 81.32 .33 | 36.6 | 8.7 | -72.1 | -26.8 | 0.0 | 10.0 |
| 108.4311 | 65.5 | 11.8 | -72.1 | 5.3 | 1.8 | 10.0 |
| 135.5389 | 43.3 | 12.2 | -72.1 | -11.6 | 0.3 | 10.0 |
| 162.6466 | 47.6 | 19.3 | -72.1 | -5.2 | 0.6 | 10.0 |
| 189.7544 | 40.1 | 18.4 | -72.1 | -13.6 | 0.2 | 10.0 |
| 216.8622 | 40.1 | 16.7 | -72.1 | -15.3 | 0.2 | 10.0 |
| 243.9677 | 47.1 | 17.0 | -75.1 | -6.0 | 0.5 | 10.0 |
| 2.71 .0777 | 34.8 | 17.3 | -72.1 | -20.0 | 0.1 | 10.0 |

E.TI: ns: 01

EI. ITR EIITTRONIC INRTNFERTME I:IL. DATA :BATIE

Test Distance: 75 ft. Azimuth : 280 degrees
Corrections based on a field decay exponent of 1.95


| 27.1038 | 38.8 | 11.0 | -72.1 | -22.3 | 0.1 | 0.0 |
| ---: | ---: | ---: | ---: | ---: | ---: | ---: |
| 54.2076 | 25.0 | 13.1 | -72.1 | -34.0 | 0.0 | 10.0 |
| 81.3114 | 28.7 | 8.7 | -72.1 | -34.7 | 0.0 | 10.0 |
| 108.4152 | 65.6 | 11.8 | -72.1 | 5.4 | 1.9 | 10.0 |
| 135.5190 | 43.8 | 12.2 | -72.1 | -11.1 | 0.3 | 10.0 |
| 162.6227 | 47.6 | 19.3 | -72.1 | -5.2 | 0.6 | 10.0 |
| 187.7285 | 45.3 | 18.4 | -72.1 | -8.4 | 0.4 | 10.0 |
| 216.8303 | 42.3 | 16.7 | -72.1 | -13.0 | 0.2 | 10.0 |
| 243.9341 | 43.9 | 17.0 | -72.1 | -3.2 | 0.5 | 10.0 |
| 271.0379 | 29.7 | 17.3 | -72.1 | -25.1 | 0.1 | 10.0 |

# FTV: م.70: <br>  DMTA PAME. 

THES MANIIFACTIJRFIS MODEL * S/N DATE TESTED

ГRCR PART $18 D$ INDUSTRIAI IITATTNG FIJITPMINT
B
PROTOTYPE NTUEMEER 11, 15E3

Test Distance: 75 ft : Azimuth : 30 n degrees Corrections based on field decay exponent of 1.95


| 27.0994 | 35.9 | 11.0 | -72.1 | -25.2 | 0.1 | 0.0 |
| ---: | ---: | ---: | ---: | ---: | ---: | ---: |
| 54.1989 | 26.7 | 13.1 | -72.1 | -32.3 | 0.0 | 10.0 |
| 81.2983 | 30.4 | 8.6 | -72.1 | -33.1 | 0.0 | 10.0 |
| 108.3978 | 65.5 | 11.8 | -72.1 | 5.3 | 1.8 | 10.0 |
| 1.35 .4972 | 45.2 | 12.2 | -72.1 | -14.7 | 0.2 | 10.0 |
| 162.5967 | 47.4 | 19.3 | -72.1 | -5.4 | 0.5 | 10.0 |
| 189.6961 | 42.1 | 18.4 | -72.1 | -11.6 | 0.3 | 10.0 |
| 216.7955 | 44.0 | 16.7 | -72.1 | -11.3 | 0.3 | 10.0 |
| 243.8950 | 49.0 | 17.1 | -72.1 | -6.1 | 0.5 | 10.0 |
| 370.9944 | 34.2 | 17.3 | -72.1 | -20.6 | 0.1 | 10.0 |

f.Tr RTOI
E. ITF EI.F RTRONIC ENT.TNIFRTHS: r.O.

DATA BASE.


27.1115
54.2231
81.3346
108.4462
1.35 .5577
162.6692
139.7308
216.8923
2.44 .0039
271.1154

| 47.7 | 11.0 | -72.1 | -13.2 | 0.2 | 0.0 |
| ---: | ---: | ---: | ---: | ---: | ---: |
| 29.2 | 13.1 | -72.1 | -29.8 | 0.0 | 10.0 |
| 34.9 | 8.7 | -72.1 | -28.5 | 0.1 | 10.0 |
| 67.1 | 11.8 | -72.1 | 6.9 | 2.2 | 10.0 |
| 4.5 .2 | 12.2 | -72.1 | -14.7 | 0.2 | 10.0 |
| 53.0 | 19.3 | -72.1 | 0.2 | 1.0 | 10.0 |
| 32.1 | 13.4 | -72.1 | -21.6 | 0.1 | 10.0 |
| 40.7 | 16.7 | -72.1 | -14.6 | 0.2 | 10.0 |
| 39.9 | 17.0 | -72.1 | -15.2 | 0.2 | 10.0 |
| 42.8 | 17.3 | -72.1 | -12.0 | 0.3 | 10.0 |

chpeked by:

PTr. 8:311
FI TTH ELTRTRONIC. FNGITNFESTN: R:O. DATA :'AIME.

TEST MANIJFACTTIJRFIR MODEL G/N DATE TESTED

FCR FART IOS INDUSTEJAI. MFATTNG FDUIDMFNT B
PROTOTYPE NOUEMBER 11, 1983

Test Distance: 75 fi. AziAuth: 340 degrees Corrections based on a field decay exponent of 1.95

27.0977
54.1995
81.2992
108.3990
135.4787
162.5985
189.6982
216.7979
243.8977
270.9974
5.3 .1
30.3
37.5
69.6
52.7
54.9
45.
$40.4 \quad 16.7 \quad-72.1$
3h. $6 \quad 17.0 \quad-72.1$
$41.1 \quad 17.3 \quad-72.1$
-R. 1
$-28.7$
$-23.7$
9.4
$-7.2$
2.1
$-8.0$
$-14.9$
$-18.5$
$-13.7$
0.1
0.0
$0.0 \quad 10.0$
0.110 .0
$2.9 \quad 10.0$
0.410 .0
1.310 .0
0.410 .0
0.210 .0
$0.1 \quad 10.0$
$0.2 \quad 10.0$
ground rf field measurements - machine C


Figure B-7. Mochine C Ground Determined Decoy Exponent



Figure B-8. Machine C Operating Frequency Fiald Intensity at 1000 feet


Figure B-9. Machine C Field Intensity vs. Frequency

1r...
Man. l. fil illii. :i mпй... \#
$:, / N$
DAIF TFSTE゙D : OCTOERR 13, 19E?
fa今t Distance: 200 ft. Aziallth : I) degrans
Corrections baseci on a field decay exponent ar j. gis



|  |  |  |  |  |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
|  | intr RAg nBud | Ant． far． nt： | Dist. <br> corr <br> Hft | Tntal dFivU／m か）1ヶilo | $\begin{aligned} & \text { rotal. } \\ & \text { uviri } \\ & \text { of } 1 \text { al } 1: \end{aligned}$ | Liait uい！m G Inille |
| 27.1734 | 36） 3 | 11.0 | －5．4 | $\cdots 7.8$ | 0.4 | 13.0 |
| 54．3469 | 13．5 | 13.0 | －55． 4 | －28．9 | 0.0 | 10.0 |
| 31．5013 | 30.5 | 0.7 | －5．5． 4 | － 16.3 | 0.8 | 10.0 |
| 108：．6938 | 84.1 | 11.5 | －5\％， 4 | $\cdots 19.5$ | 0.1 | 10.0 |
| 1．35． 16.75 | 3：7．2 | 13． 3 | －65， 4 | 16， 11 | 0.7 ？ | 111.6 |
| 16.3 .0467 | 4E， 4 | 15.4 | －56． 4 | 9.3 | $\therefore \%$ | 10.0 |
| 1\％11．？141 | ［3．6． 4 | 13．， 5 | －5\％ 4 | $-10.7$ | 0.3 | 10.0 |
| 2：7．3876 | 41．5 | 15．． 7 | －5io－ 4 | 2．月 | 1.4 | 10.0 |
| 944． 5610 | i2；， 7 | $1 \%$ ， | 5．7， 4 | －15．7 | 0.12 | 10.0 |
| $971.734 \%$ | 29．3 | 17．3 | －E5：， 4 | －8．\％ | 0.4 | 10.0 |

dFiuU/M uViri UVim

$\therefore$ :N : : OPDTOTYBE DATE TESTED : ORTOAFR 13, 14R?
 Comections based ori a ripld deray exponent of 1.55


| ?7. $17: 3$ | 34.5 | 11.0 | -5\% . 4 | $-9.9$ | 0.3 | 13.0 |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 54.3528 | 13.2 | 13.0 | -55, 4 | -29.7 | 0.0 | 10.0 |
| 31.5099 | 23. 1 | 8.7 | -55. 4 | -1E.7 | 0.1 | 10,0 |
| $10: 5051$ | 17.9 | 11.9 | -55.4 | $-6.7$ | 0.1 | 10.0 |
| 1.5.1.533 4 | S. 1 | 1:3.3 | -5,5, 4 | . 2 Cl . 1 | 0.1 | 10.0 |
| 167.0577 | $4 \% .0$ | 19,4 | -5.5, 4 | 8.9 | - 5 | 10.0 |
| 1911. 2.340 | A: , i? | 18.3 | ..-5.5. 4 | -74.7 | 0.7 | 10.0 |
| 21\%.4103 | 4\%.0 | 16,7 | - Es, 4 | 3.3 | 1. ${ }_{1}$ | 10.0 |
| ? 44.510 ma | 3.4 | 17.7 | -5, 4 | $-13.5$ | 0.:? | 10.0 |
| 271.7609 | 20.6, | 17.3 | - 5゙心 4 | $-17.6$ | 0.1 | 10.0 |


MOni : * : C
S/N : : RROIGTYME
DATE TESTITD : URTOENK 13, 190\%
Corrections hased on a field deray exponent or 1.5 s





Mintit:C
$\therefore$ : PA PROTOTYPE
DATE TESTED : OCTOAFR 1?, 19C3




FT．A．




Mll！！：C
$\therefore$ ：iv ：PROTOTYBE
DAIE TESTLD ：GCTMHTR 13， $15 \Omega 3$
rest Distancm： 200 ft．Asinuth：1：？degreas Corrections based on a field decay exponent of $1.9 \%$

|  | Mtrerdg | Ant． | Dist． | T0tal | rotal | Limit |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
|  |  | $f a c$ ． | corr | dTuV／n | UV／M | いV／A |
| Mitr | d！3uV | dL | Ht | ग 1milo | （1 MIJ．cis | C 1mile |


| ？ 7 ．133．3 | 32． 4 | 11.0 | －－5．5．5 ． 4 | $-12.0$ | 0，23 | 0.0 |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 54.3647 | 13.5 | 13.0 | －55． 4 | －－24．0 | 0.0 | 10.0 |
| U1，6470 | 31.0 | 8.7 | －55． 4 | $-15.3$ | 0． 3. | 10.0 |
| 10ヶ．729こ | 78．6 | 11.9 | －－5．4 4 | －15．0 | 0.2 | 10.0 |
| ：35．9116 | P6．7 | 1：3．3 | －55．4 | －16．5 |  | 10，0 |
| 16－5．0940 | 44.7 | 19.4 | －5E， 4 | 10.7 | 7． 4 | 10， 0 |
| 1713．2\％ | 34.7 | 13．3 | －5．5． 4 | $\cdots 2.4$ | 0.7 | 10.0 |
| 217．4581． | 43.0 | 16.7 | －55．4 | 4．3 | 1.4 | 10.0 |
| 1？44．6110 | 2\％．7 | 17．0 | －－55． 4 | －19．5 | 0.6 | 11.0 |
| 278．8933 | 37.6 | 17.3 | $-55.4$ | －0．t． | 0.9 | 10.0 |





Corrections based on arield decay exponprit of 1.95

| :rap. | Mtr Rat | Ant. <br> far. | Diat. corr | Tital. aFいい/M | rotal uU/M | Limit UV/i: |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| Mit: | 48300 | His | H? | ¢ Imile | \% 1 MiJ . 0 | e Imile |


| 7.7.17? | 31.7 | 11.0 | -55. 4 | -12.7 | 0.2 | 0.0 |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 54.3851 | 14.5 | 13.0 | -55. 4 | -2ก.0 | 0.0 | 10.0 |
| ก1.5777 | 30.4 | 13.7 | -55. 4 | --16.4 | 0.2 | 10.0 |
| 105:.7702 | 27.4 | 11.9 | -56.4 | -16.? | 0.2 | 10.0 |
| 1.35. 76.30 | 32.6 | 12.3 | -55, 4 | $-\mathrm{O} .5$ | 0.3 | 10.0 |
| 163.1553 | $4 i .5$ | 19.4 | -56. 4 | 10.5 | 3.3 | 10.0 |
| $1715.347 \%$ | 39.9 | 13.3 | $5 \% .4$ | 7.3 | 0.4 | 10.0 |
| 217.5404 | 43.5 | 16.7 | --55. 4 | 4.81 | 1.7 | 10.0 |
| 244.7 .350 | 31.7 | 17.0 | -55. 4 | 6.7 | 0.5 | 10.0 |
| 271.985 | 32.8 | 17.3 | -55. 4 | -5. 4 | 0.5 | 10.0 |



Pimult，Fl，l：1A：：
m（7）HEL
：$/ \mathrm{N}$
DATE TESTFD ：OCTOFIFR $13,1 \% \boxed{3}$

Test Distance： 200 ft．Azimilt ； 200 degrepes Corrections hased on a fifld der：ey exponent of 1.95

| Fref． | Mir Relg | Ant． <br> fac． | Dist． <br> corr | Tッチッ！ dEuV／m | Total uV／m， | Limit uU／n |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| itil？ | jund | Hs | 14 | （3）Imila | Q 1mila | E laile |


| 3．7．18B：3 | 28.8 | 11.0 | ．55． 4 | －17．6 | 0.1 | 0．0 |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 54.3776 | 14．1 | 12．0 | －sis． 4 | －20．4 | 0.0 | 10.0 |
| 31.5684 | 3.7 ， | 0.7 | －55． 4 | －17．？ | 0.1 | 10.0 |
| 105．7552 | 2h． 3 | 11.9 | －Fin， 4 | －17．3 | 0.1 | 10.0 |
| 1．3：5， 7441 | ． 30.4 | 12．3 | ． 5.5 .4 | －12． 3 | 0.3 | 10.0 |
| 16\％．1328 | 42． 4 | 19.4 | －55．4 4 | 1.4 | 2.1 | 10.0 |
| 170．3816 | 33.7 | 18.3 | －5，3．4 | 8.1 | 0.4 | 10.0 |
| 217．5104 | 41.2 | 14.7 | －5ES． 4 | C．${ }^{\text {c }}$ | 1.3 | 10.0 |
| 244．67\％？ | 2．7．7 | 17．17 | －5， 4 | －8．7 | 0.4 | 10.0 |
| 271．88E0 | 36.4 | 17.3 | －55．4 | $-1.8$ | 0.8 | 10.0 |




Test Distance: ? 00 fr. Azimuth : 241 degrees Corrections based on a field decal pxponery or 1.95



rest Distanc: : Soif ft. Azimuth i 2(,l) degraes Corrections hosed on a field decay exporient of 1.95




- 1: $\quad \therefore$ :
E. © i. TiNiti:.
i月ii. A

Mift: : . : 11.si:
MīidL.
C
?ROTOTYPE
G/fi
DATE TEMTID : OCTOLCK 13, $198, ?$
Test Distanre: $20 n$ ft. Atimuth 3010 gegrapes Corrections based on a field decay exponent of 1.95



FTn: i., $8 \%$

Baja - manion

Test Distance: $\quad$ : OOn ft. Azimuth : 3ann degrees
Corrections tiasect on field decay expanent of 1.95






```
MONTL * : C
:IN
FROTUTYPE
DATE TESTFD : ORTTEERK 13, 1%AS
```

rert Distanre: ixn fr. Azinuth: 340 degreses
Corrections based on a field decay exponent of 1.95

| Fred. | Mer Relis | Ant. | Dist. | Total | Total | Limit |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
|  |  | fac. | corr | dEwU/m | uV/a | uV/ri |
| ifilz | AT:MV | 10 | HT | ¢ 1ヵile | m 1mila | e Imile |


| 2.7.1751 | 37.1 | 11.0 | -5.5. 4 | -.5. 3 | 0.5 | 11.0 |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 54.3503 | 16.0 | 13.0 | -55. 4 | -26.4 | 0.0 | 10.0 |
| 81.58 .54 | 33.3 | 1.7 | -55. 4 | -13.5 | 0.2 | 10.0 |
| 100: 7005 | 39.1 | 11.5 | -5E, 4 | -14.5 | 0.2 | 10.0 |
| 1.35.13.757 | 23.10 | 12.33 | -.5.5. 4 | -15.2 | 0.3 | 10.0 |
| 163.0508 | 43.4 | 19.4 | -55. 4 | 7.3 | 2.3 | 10.0 |
| 190. 3 , 57 | 27.7 | 18.3 | -55. 4 | -7.2 | 0.3 | 10.0 |
| 217.4011 | 43.6 | 16.7 | -5E. 4 | 4.9 | 1.8 | 10.0 |
| 244.5.7A"? | 2¢.0 | 17.11 | -5\%.4 | -12.4 | 0.8 | 10.0 |
| 271.7512 | 33.5 | 17.3 | -55.4 | -4.7 | 0.6 | 10.0 |



Figure B-10. Machine D Ground Defermined Decay Exponent


Figure B-11. Machine D Operating Frequency Field Intsinsity
at 1000 feet


Figure 8-12. Machine D Field Intensity vs. Frequency

T:.
 MODEL *
!iN DATE TESTED

D
PROTOTYPE
DCTOFEK 13, 19E.3

Test Distance: 200 ft. Arimyth: 0 degrees Corrections based on field decay exponent of 1.95



そ7: fioll



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$\therefore \mathrm{CH}$
rROTOTYEE
DATE TESTFD : NC:TOFFRR 17, 1993
 Corrections based on a fiejd decay exponent of 1.95


2:3. 186.3
54.3726

B1. 55.38
108:.7451
135.7314

16?.1177
190.30411
217.4902
3.4 .4 .676 .5
;71.86"と
39.7
14.9 27.:3 22. 8 19.7 40.8 ? 6.7 $40.5 \quad 16.7 \quad-55.4 \quad 1.8 \quad 1.8 \quad 10.0$ $\begin{array}{llllll}31.0 & 17.9 & -5.5 .4 & -17.4 & 0.1 & 11.0 \\ 27.6 & 17.3 & -55.4 & -6.8 & 0.4 & 10.0\end{array}$

Mridiji Gr. IUR.

| M(IJFL | : D |
| :--- | :--- |
| $: G / N$ | PROTOTYPE |
| DATE TESTED | : OCTOERR 13,1983 |

rest Distance: 200 ft. Azimuth : 40 degrees Corrections based on a field decay exponent of $1 . \mathrm{g}^{2}$


| 2.7 .174 .3 | 36.1 | 11.0 | -55.4 | -8.3 | 0.4 | 0.0 |
| ---: | ---: | ---: | ---: | ---: | ---: | ---: |
| 54.3487 | 12.9 | 13.0 | -55.4 | -29.5 | 0.0 | 10.0 |
| 31.5230 | 30.6 | 8.7 | -55.4 | -16.2 | 0.2 | 10.0 |
| $118: .6974$ | 17.2 | 11.9 | -55.4 | -26.4 | 0.0 | 10.0 |
| 135.8717 | 16.9 | 12.3 | -55.4 | -26.3 | 0.0 | 10.0 |
| 16.3 .0461 | 47.0 | 19.4 | -55.4 | 51.5 | 2.0 | 10.0 |
| 171.2 .04 | 14.6 | 18.3 | -55.4 | -13.5 | 0.1 | 10.0 |
| 217.3948 | 41.8 | 16.7 | -55.4 | 3.1 | 1.4 | 10.0 |
| 244.5691 | 25.9 | 17.1 | -55.4 | -12.5 | 0.2 | 10.0 |
| 271.7435 | 26.4 | 17.3 | -55.4 | -11.8 | 0.3 | 10.0 |




$$
\begin{aligned}
& \text { ETE: } \because 11
\end{aligned}
$$

$$
\begin{aligned}
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\end{aligned}
$$



M(ID)FL: $: ~ D$
S/N : PROTOTYBE
DATE TESTED : OCTOFIRR 13, 1983
Test Distance: 200 ft. Atimuth : 101 degreps
Corrections based on a field decay expanent of 1.9 k

27.1703
54.3417
81.5185

108:.6833
$1.3 .5 .8: 541$
163.0250
$170.185: 3$
217.3666
244.5374
271.7083
36.4
13.4
26.13
25.0
3.6 .4
40.0
32.1
39.0
24.2
33.6

| 11.0 | -5.5 .4 |
| ---: | ---: |
| 13.0 | -55.4 |
| 8.7 | -55.4 |
| 11.9 | -55.4 |
| 1.3 .3 | -55.4 |
| 19.4 | -55.4 |
| 18.3 | -5.5 .4 |
| 16.7 | -55.4 |
| 17.0 | -55.4 |
| 17.3 | -55.4 |

-8.0
-2.9 .0
-20.0
-18.6
-16.0
4.7
-5.0
0.3
-14.2
-4.6

| 0.4 | 0.0 |
| ---: | ---: |
| 0.0 | 10.0 |
| 0.1 | 10.0 |
| 0.1 | 10.0 |
| 0.1 | 10.0 |
| 1.7 | 10.0 |
| 0.6 | 10.0 |
| 1.0 | 10.0 |
| 0.72 | 10.0 |
| 0.6 | 10.0 |

r:herckert

```
                        #: %!%11
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\begin{aligned}
& \text { Test Disiance : } 200 \text { it.inuth : } 1 \text { an degrees } \\
& \text { Corrections based on a field deray exponerit of } 1.92
\end{aligned}
$$




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\begin{aligned}
& \because \because \therefore \therefore 11
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$$

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 MIIDFL * $\therefore 3 / N$ DATE TESTED

D
PROTOTYPE
OCTOEER 13, 1983

Test Distanre: 200 ft. Arimuth i 141 degreps Corrections based on field decay exponent of 1.95



$$
\begin{aligned}
& \text { ETP:0.11 }
\end{aligned}
$$



Test Distance: 200 ft. Azimuth: 1ho degrepes Corrections based on tield deray exponent of 1.95

| $\because \mathrm{rer}$ | Mtr RAg | Ant. | Digt. | Tot.71 | Total. | Linit |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
|  |  | fac. | corr | dSuU/m | UV/M | UV/M |
| AlH: | ATuV | A5 | dE | 日 tmilm | e 1mile | E lmile |


| 27.1785 | 2;3. 4 | 11.0 | -55.4 | -22.0 | 0.1 | 0.0 |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 54.3569 | 13.5 | 13.0 | -55.4 | -28.9 | 0.0 | 10.0 |
| 81.8 .754 | 27.8 | 6.7 | -55.4 | $-17.0$ | 0.1 | 10.0 |
| 108.7139 | 26.7 | 11.9 | -55.4 | $-16.9$ | 0.1 | 10.0 |
| $13: 5.1392 .3$ | 31) 11 | 12.3 | -5.5. 4 | -14.? | 0. ${ }^{2}$ | 11). 0 |
| 163.0708 | 43.6 | 19.4 | -55.4 | 6.6 | 2. 1 | 10.0 |
| 170.:2493 | 27.6 | 18.3 | $-.55 .4$ | -7.5 | 0.4 | 10.0 |
| 217.4277 | 4. . 7 | 16.7 | -55. 4 | 4.0 | 1.6 | 10.0 |
| 244.604.? | 29.8 | 17.0 | -55.4 | $-3.6$ | 0.4 | 10.0 |
| 271.7847 | 33.5 | 17.3 | -55.4 | -4.7 | 0.6 | 10.0 |

ETB: :\%11




「est Distanc: : 20 fot. Amuth : 1 Bn degreps Corrections hased on a field decay exponent of 1.95

Freq. MtrRAg Ant. Dist. Tntal Lotal. Limit
Mitr atrind
fac. corr
तIS AE
dFuV/m uU/m UVim

27.13513
54.3717
81.5.575
$10 \mathrm{f:} 7434$
1.3:5. 9272
163.1150 195.3007 217.4867
 271.8584
17.7
12.0
30.0
35.3
;R. 17
43.0
2.4.13
44.3
?? 7
33.5
11.0
$-55.4$
-55. 4
-. 5.5 .4
-55. 4
..55. 4
$15.4 \quad-55.4$
$\begin{array}{ll}18.3 & -55.4 \\ 16.7 & -55.4\end{array}$
$\begin{array}{ll}17.0 & -5.5 .4 \\ 17.3 & -55.4\end{array}$
$\begin{array}{ll}17.0 & -55.4 \\ 17.3 & -55.4\end{array}$
13.0
8.7
11.7
15.3
17.3
-24.7
-30.5
-16.18
-18.3
-16.3
7.0
-12.3
5.6
-13.7
-4.7
0.1
0.0
$-4.7$
0.
10.0
4
$-30.5$
0.0
10.0
-

F!r ؛..|l



Mats:l: Ai lls...
MUDTL : D
:I/N : PROTGTYPE
DATE TESIIED : OCTCIRIRR 13, 198:3
Tert Distanrif: 200 ft. Azimuth: 20ll degrepes Correctinns baser on field decay exponent of 1.9 in



$$
\begin{aligned}
& \text { ©: } \because 41
\end{aligned}
$$

$$
\begin{aligned}
& \text { (1.n): in iA!.: }
\end{aligned}
$$


 MCIMFL $\#: D$ S/N

PROTOTYPE DATE TESTFD : OCTOFER 13, 198.3

Test Distance: 200 ft. Azimuth : 220 degrees Corrections based on field decay exponent of 1.95

| Frant. | Mtr RAJ | Ant. <br> fac | Dist. corr | TnTal dFuU/M | Total uV/m | Limit uV/M |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| Mitz | dEuV | dB | AB | ค 1mith | O IMİm | Q 1mile |


| 27.18.5.5 | 31.4 | 11.0 | ..55.4 | $-13.0$ | 0.2 | 0.0 |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 54.3710 | 13.2 | 13.0 | -55.4 | -29.3 | 0.0 | 10.0 |
| 81.5566 | 27.1 | 8.7 | -55.4 | -17.7 | 0.1 | 10.0 |
| 108.7421 | 26.2 | 11.9 | -55. 4 | -.17.4 | 0.1 | 10.0 |
| 1.35.927i | 3.7.2 | 12.3 | .-55. 4 | -14.0 | 0. $\mathrm{i}^{3}$ | 10.0 |
| 163.1131 | 41.5 | 19.4 | -55.4 | 6. 5 | 1.9 | 10.0 |
| 191.2937 | 2.4.0 | 113.3 | -5.5. 4 | -B. 1 | 0.4 | 10.0 |
| 217.4843 | 36.0 | 16.7 | -55. 4 | -2.7 | 0.7 | 10.0 |
| 244.6677 | 2? 3 | 17.0 | -55. 4 | -15.7 | 0.8 | 10.0 |
| 271.85ち? | 32.6 | 17.3 | -55. 4 | $-5.6$ | 0.5 | 10.0 |


 Corrections based on field decay exponent of 1.5E

-: hanc: kral



MANIIF frifillis: be :

MCHIL * BiN DATE TESTFD : DR.TGIER 1: 1983

Test Distarict : 200 ft. Arimuth : 2fill degrees Corrections based on a field decay exponent of 1.95


| 27.1832 | 40.6 | 11.0 | -55.4 | -3.8 | 0.6 | 0.0 |
| ---: | ---: | ---: | ---: | ---: | ---: | ---: |
| 54.3665 | 13.2 | 13.0 | -55.4 | -29.3 | 0.0 | 10.0 |
| 81.5497 | 24.9 | 8.7 | -55.4 | -21.9 | 0.1 | 10.0 |
| 108.7330 | 28.0 | 11.9 | -55.4 | -15.6 | 0.2 | 10.0 |
| $13: 9.9167$. | 17.4 | 12.3 | -.55 .4 | -2.3 .13 | 0.1 | 10.0 |
| 163.0995 | 35.0 | 15.4 | -55.4 | -3.0 | 0.7. | 10.0 |
| 170.8 .9 .7 | 20.13 | 13.3 | -55.4 | -16.3 | 0.2 | 10.0 |
| 217.4660 | 34.8 | 16.7 | -55.4 | -3.7 | 0.8 | 10.0 |
| 244.6492 | 17.3 | 17.0 | -55.4 | -21.1 | 0.1 | 10.0 |
| 271.83 .5 | 24.1 | 17.3 | -55.4 | -14.1 | 0.2 | 10.0 |

$$
\begin{aligned}
& \because \because: \because 11
\end{aligned}
$$


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MTIDFL : D
$\therefore / N$ : RROTOTYPE
DATF. TESTTD : CCTOFFK 13, 198!3
Test Distancr: 200 fi. Azimuth : 30t degrepes
Corrections based on a fiejd decay expanent of 1.95




$$
\text { ETK }: \because,
$$



rest Distance: 200 ft. Azimuth : 340 degrepes Corrections based on field decay fxponent af 1.95

Freq. Mtr RAg Ant. Dist. Intal rotal Linit fac. corr $\quad d F_{\| V} \quad \mathrm{VV} / \mathrm{M} \quad \mathrm{VV} / \mathrm{m}$ MHz druju dif timite e lmile e imile


## Appendis C.

Details of the calfbration of an EMCO 3104 biconical antenna are repot:ed. Calibration data for this antenna at 27 MHz and 109 Miz are


```
Details of the calibration of an EMCO
3104 biconical antenna are reported.
Calibration data for this antenna at 27
MHz}\mathrm{ and 109 MHz are given.
```

```
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    Avionics Engineering Center
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    Oh;o University
Athens, Ohio 45701
```

Apr $\$ 11984$

## Prepared for

Federal Aviation Administration Spectrum Engineering Division, AES-500 Washington, D.C. 20590

Contract No: DTFAO 1-83-C-10007
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1. PURPOSF:1
II. THEOYY OF OPERATION ..... 2
III. EQUIPMENT ..... 4
IV. SETUP ..... 5
V. PROCEDURES ..... 6
VI. EXAMPLE ..... 7
VII. REFERENCES ..... 9
VIII. ACKNOWLEDGEMENTS ..... 10

## I. PIIRPOSE

To calibrate a biconical antenna at 27 and 109 MH\% so that the antenna may be: used to tratsmit a calibrated RF fleld or measure an unknown kF field. This procedure yields the absolute fitin over an isotropic source for an antenna, which can be used to determine an antenna factor to use for measuring purposes.
II. THEORY OF OPERATION

This calibration procedure is based on the material presentid in [1] and is summarized below.

The absolute gain of an antenna over an tsot ropic source ran be determined if there are three relatively similar an ennas (that is three antennas with similar radiating patterns). This me hod is based on the relation of the product of two antenna gains to the recuived and transmitted power given by equation $l$ when the two antennas are set up so that one is transmitting to the other.[2]

$$
\begin{equation*}
G=\sqrt{G_{01} G_{02}}=\frac{4 \pi s}{\lambda} \sqrt{\frac{W r}{W t}} \tag{eq.1}
\end{equation*}
$$

where $G_{01}$ is the gain of antenna 1
$G_{02}$ is the gain of antenna 2
$S$ is the spacing between the two antennas
$\lambda$ is the wavelength
$W_{r}$ is the received power, and
$W_{t}$ is the transmitted power
If three antennas are used then there are three possible combinations of two antennas and three gain measurements. From these three measured product gains the gain of each antenna can be calculated as follows:

If the test configurations are
Transmit Receive

| no. 1 | Ant. 1 | Ant. 3 |
| :--- | :--- | :--- |
| no. 2 | Ant. 2 | Ant.3 |
| no. 3 | Ant. 2 | Ant. 1 |

and
$G_{0 n}=$ isotropic gain of antenna $n$
$G_{n}=g a i n$ of nth test configuration
$W_{r n}=$ received power of nth configuration
$W_{t n}=$ transmitted power of nth configuration
then

$$
\begin{align*}
& G_{1}=\sqrt{G_{01} G_{03}}=\frac{4 \pi S_{1}}{\lambda_{1}} \sqrt{\frac{W_{r 1}}{W_{t 1}}}  \tag{A}\\
& G_{2}=\sqrt{G_{02} G_{03}}=\frac{4 \pi S_{2}}{\lambda_{2}} \sqrt{\frac{W_{r 2}}{W_{t 2}}} \tag{B}
\end{align*}
$$

$$
\begin{equation*}
c_{3}=\sqrt{c_{02} G_{01}}=\frac{4 \pi S_{3}}{\lambda_{3}} \sqrt{\frac{W_{r 3}}{\omega_{t 3}}} \tag{C}
\end{equation*}
$$

By as suming that $S 1=S 2=S 3=S$ and $\lambda_{1}=\lambda_{2}=\lambda_{3}=\lambda$ which can be achieved by using identical test configurations.
from (A) $G_{01} G_{03}=\left(\frac{4 \pi S}{\lambda}\right)^{2} \frac{W_{r 1}}{W_{t 1}} \rightarrow G_{03}=\left(\frac{4 \pi S}{\lambda}\right)^{2} \frac{W_{r 1}}{W_{t 1}} \frac{1}{G_{01}}$
from (B) $G_{02} G_{03}=\left(\frac{4 \pi S}{\lambda}\right)^{2} \frac{W_{r 2}}{W_{t 2}} \rightarrow G_{03}=\left(\frac{4 \pi S}{\lambda}\right)^{2} \frac{W_{r 2}}{W_{t 2}} \frac{1}{G_{02}}$
equating (Al) and (BI) gives

$$
\begin{equation*}
G 2=\frac{W_{t 1} W_{r 2}}{W_{r 1} W_{t 2}} G_{01} \tag{D}
\end{equation*}
$$

substituting (D) into (C) yields

$$
\begin{equation*}
G_{01}=\frac{4 \pi S}{\lambda} \sqrt{\frac{W_{r 1} W_{t 2} W_{r 3}}{W_{t 1} W_{r 2} W_{t 3}}} \tag{E}
\end{equation*}
$$

substituting (E) into (D) yields

$$
\begin{equation*}
G_{02}=\frac{4 \pi S}{\lambda} \sqrt{\frac{W_{t 1} W_{r 2} W_{r 3}}{W_{r 1} W_{t 2} W_{t 3}}} \tag{F}
\end{equation*}
$$

and substituting (F) into (BI) gives

$$
\begin{equation*}
G_{03}=\frac{4 \pi S}{\lambda} \sqrt{\frac{W_{r 1} W_{r 2} W_{t 3}}{W_{t 1} W_{t 2} W_{r 3}}} \tag{G}
\end{equation*}
$$

Thus, ( $E$ ), ( $F$ ), and ( $G$ ) are expressions for the absolute gains of antennas 1,2 , and 3 respectively.

The conversion to antenna factor from power ratio gain in $d B$ can then be calculated using equation 2. This antenna factor is then added to a voltmeter reading to obtain the absolute signal strength in $\mathrm{dBuV} / \mathrm{m}$.

$$
\text { for } \begin{aligned}
K & =20 \log (f)-G-29.8 \\
Z & =50 \text { ohm }
\end{aligned}
$$

(eq. 2)
where $K=$ antenna factor in $\mathrm{dBuV} / \mathrm{m}$
$f=$ frequency in $M H Z$
$G=$ antenna power ratio gain in $d B$
III. EQUIPMENT

```
Antenrias
    Biconical antenna, EMCO model 3104, 0.ll. ro. 1484
    Dipole antenna CU-683/URM-7, 0.U. no. 037"
    Dipole antenna marked 'EMI REFERENCE'
    11 antenna element:s AB-21/GR
    2 antenna elements AT-848/URM-7
    2 aluminum antenna elements 40 inches lorg
Signal generator
    Wavetek 3000 - O.U. no. 1298
    Avantek RF power amplifier
Detection units
    EMC-25 Selective voltmeter
    HP141T Spectrum analyzer
Directional Coupler
    HP778D Dual Directional coupler serial no. 1144A04704
Antenna towers
    Clark tower - max. height approx. 70 feet
    Tripod stand MT-1947/URM-7 - max. height 15 feet
    Tripod stand TRP-25, 0.U. no. 1483
Cables - all cables to be 50 ohm coaxial
    EMI cable A (approximately 35 feet long)
    EMI cable B (approximately 80 feet long)
    Several short interconnect cables
DC power supply
        HP 6237B triple output
AC power source
        gas powered alternator
        1 0 0 \mathrm { ft } \text { . extension cord}
        multiple outlet extension cord
Connectors for all setups
```

[V. SETUP
The entire tesit setup is to be located in a place that is as free from RF noise as possible and clear of any large metallic objects that may in any way alter the propagation of the transmitted signal. It is suggested that any large metallic objects in the test area be at a distance from either antenna equal to not less than three times the spacing between the two antennas. Also the area chosen should be as flat as possible and the surface should be of approximately the same material throughout the test area.

The two antennas are to be separated by a distance such that the receiving antenna is in the 'far field' of the transmitted signal. This distance is to be a minimum of three times the wavelength.[3] In addition, the two antennas are to be oriented for maximum coupling and placed on towers at heights such that the summation of the direct wave and the ground reflected wave is a maximum.

The height requirement of the antenna setup is that the two antennas be at heights that cause the ground reflected wave present at the receiving antenna to be in phase with the direct wave, so that a maximum signal is received. The height requirement is due to the fact that near the point of maximum combined signals the variation in signal strength with height is at a minimum, thus giving a more uniform field. Although the point where the antenna must be placed has to be determined by moving the antenna vertically and watching the received signal for a maximum, a simplified formula that gives antenna height $h_{2}$ in terms of antenna height $h_{1}$, spacing $S$, and the wavelength is given in equation 3.[4]

$$
h_{2}=\frac{\lambda}{4} \times \frac{S}{h_{1}}
$$

$$
\begin{aligned}
& \text { with } \quad \begin{aligned}
& n=0,2,4, \ldots \text { for minimums and } \\
& n=1,3,5, \ldots \text { for maximums } \\
& \text { providing } S>h_{1} \text { or } h_{2}
\end{aligned} .
\end{aligned}
$$

The signal generator Avantek amplifier combination is used as the source for the transmitting antenna and the HP141T spectrum analyzer is used to measure the received signal. The output of the RF amplifier is fed to the transmitting antenna through the dual directional coupler. The dual directional coupler is used so that the forward and reflected power to the transmitting antenna can be measured with the EMC-25 receiver. The transmitted power is then calculated by subtracting the reflected power from the forward power. Note that the cable attenuation must be considered when performing the measurements of forward, reflected, and received power. Also note that the value of the transmitting antenna cable attenuation is subtracted from the measured forward power and is added to the measured reflected power.

## V. Procedures

1. Set up the equipment as described in the SETUP section with the DC power supply prowiding power to the Avantek RF power amplifier. Turn on all the equipment and adjust the controls so that a signal at least 20 dB above the noise level can be detected at the receiving end. Note that a load should be applied to the $R F$ power amp before $D C$ power is applied.
2. Adjust one or both antennas in altitude and/or orfentation so that a maximum signal level of sufficient amplitude ( $>20 \mathrm{~dB}$ above noise) is detected at the receiver.
3. Record the forward, reflected, and received power. Also record the frequency setting, height of both antennas, and the spacing between antennas along with the description of the two antennas used.
4. Exchange the transmitting antenna with the antenna previously unused, keeping the height and spacing of the antennas the same (remember to turn off power to the RF power amp before disconnecting the transmitting antenna). Turn the power amp back on when the antenna is in place and adjust the signal generator setting, if necessary, to obtain proper received signal.
5. Record the information listed in part 3 for this antenna configuration.
6. Obtain the measurements for the final configuration by exchanging the receiving antenna with the antenna first used as the transmitting antenna and repeating the prucedures above.
7. Compute the gain of the antennas using the formulas presented and the transmitted and received power just measured. Remember that the transmitted power is equal to the forward power minus the reflected power (do not forget to convert from dBm to watts before subtracting).

## VI. EXAMPLE

This section describes in detail the tost sotup used on Soptember 9,1983 and the results obtained by Jim Nickum, Hill Drury, and Dave Quiact. Antenna numbers given are referenced to the configurations in the 'THEORY OF OPERATION' section.
$\frac{109}{\text { Ante }} \frac{\mathrm{MHZ}}{\mathrm{nna}}$
Antenna 1: Dipole antenna CU-683/URM-7 with AT-848/URM-7 element each side extended for antenna length equal to $1 / 2$ wavelength at 109 MHZ .

Antenna 2: Dłpole antenna 'EMI REFERENCE' with AB-21/GR element each side.

Antenna 3: Biconical antenna ENCO 3104
Separation distance: 41 feet
Receiving antenna height: 6 feet
Transmitting antenna height: 13 feet
$P_{\text {fwdl }}=-9.2 \mathrm{dBm}=120.2 \mathrm{E}-3 \mathrm{~mW}$
$P_{\text {rfil }}=-15.8 \mathrm{dBm}=26.3 \mathrm{E}-3 \mathrm{~mW}$
$W_{t 1}=P_{f w d 1}-P_{r f 11}=93.9 \mathrm{E}-3 \mathrm{~mW}=-10.3 \mathrm{dBm}$
$W_{r 1}=-44.2 \mathrm{dBm}$
$P_{f w d 2}=-9.2 \mathrm{dBm}=120.2 \mathrm{E}-3 \mathrm{~mW}$
$P_{\text {rfi2 }}=-15.8 \mathrm{dBm}=26.3 \mathrm{E}-3 \mathrm{~mW}$
$W_{t 2}=-10.3 \mathrm{dBm}$
$W_{r 2}=-45.2 \mathrm{dBm}$
$P_{\text {fwd }}=-9.2 \mathrm{dBm}=120.2 \mathrm{E}-3 \mathrm{~mW}$
$P_{r f 13}=-15.8 \mathrm{dBm}=26.3 \mathrm{E}-3 \mathrm{~mW}$
$W_{t 3}=-10.3 \mathrm{dBm}$
$W_{r 3}=-39.7 \mathrm{dBm}$
$G_{01}=2.17$
$6_{2}=1.72$
$\mathrm{G}_{3}=$ Biconical gain $=0.61$

27 MHZ
Antenna 1: Dipole antenna CU-683/URM-7 with 4 AB-21/GR elements plus AT-848/URM-7 elements each stde extended to 1/2 wavelength at 27 MHZ.

Antenna 2: Dipole antenna marked 'EMI REFERENCE' with three $A B-21 / G R$ elements plus aluminum extensions each side. Length equal to $1 / 2$ wavelength at 27 MHz .

Antenna 3: Biconłcal antenna EMCO 3104
Separation distance: 100 feet
Receiving antenna height: 63 feet
Transmitting antenna height: 13 feet
$P_{\text {fwd } 1}=-7.8 \mathrm{dBm}=166 \mathrm{E}-3 \mathrm{~mW}$
$P_{r f 11}=-19.0 \mathrm{dBm}=12.59 \mathrm{E}-3 \mathrm{~mW}$
$W_{t 1}=-8.14 \mathrm{dBm}$
$W_{r 1}=-53.5 \mathrm{dBm}$
$P_{\text {fwd2 }}=-7.8 \mathrm{dBm}=166 \mathrm{E}-3 \mathrm{~mW}$
$P_{\text {rf12 }}=-17.8 \mathrm{dBm}=16.6 \mathrm{E}-3 \mathrm{~mW}$
$W_{t 2}=-8.25 \mathrm{dBm}$
$W_{r 2}=-54.0 \mathrm{dBm}$
$P_{f w d 3}=-7.8 \mathrm{dBm}=166 \mathrm{E}-3 \mathrm{~mW}$
$P_{r f 13}=-17.8 \mathrm{dBm}=16.6 \mathrm{E}-3 \mathrm{~mW}$
$W_{t 3}=-8.25 \mathrm{dBm}$
$W_{r 3}=-33.5 \mathrm{dBm}$
$G_{1}=1.96$
$G_{2}=1.79$
$G_{03}=G a i n$ of Biconical $=0.017$

Biconical antenna factor $=16.4 \mathrm{~dB}$
[1] Kraus, John D., 'Antennas', McGraw-H111 Booi. Company, New York, 1950, pp. 448‥457.
[2] Ibさd., p. 456.
[3] "Calibration Principles and Procedures for Field Strength Meters (30 Hz to 1 GHz$)^{\prime \prime}$, National Bureau of Standards Technical Note 370, March 1969, p. 105.
[4] "The ARRL Antenna Book", American Radio Relay League, Inc., Newington, Connecticut, 1974, p. 318.

## VIII. ACKNOWI.EDGEMENTS

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