EASA

PROPOSED AIRWORTHINESS DIRECTIVE



PAD No: 07-145

Date: 31 August 2007

No person may operate an aircraft to which an Airworthiness Directive applies, except in accordance with the requirements of that Airworthiness Directive unless otherwise agreed with the Authority of the late of agistry.

Type Approval Holder's Name:

Type/Model designation :

The Boeing Company

747 series

TCDS Number: FAA A20WE

Foreign AD: This AD is related to, and prompted by NPRM Docetor. FAA-2007-28385.

Supersedure: For aircraft that operate under Electedure in the equirements of this AD take precedence over those contained in FAA AD 2 7-XX

ATA 28	Fuel System - del Implementa on	k Sy tem Airworthine	ss Limitations (AWL) -
Manufacturer(s):	Boeing Airpluse Col	n'	

Applicability:

Mod 7 (-100, 74) 100B, 747-100B SUD, 747-200B, 747-200C, 747-200F, 747-300, 74, SR, and 747SP series aircraft, all serial numbers.

Reason:

the FAA proposes to adopt a new airworthiness directive (AD) for all Boeing Model 71,100,747-100B,747-100B SUD, 747-200B, 747-200C, 747-200F, 747-300, 747-300, and 747SP series airplanes. This proposed AD would require revising the FAA-approved maintenance program by incorporating new airworthiness limitations (AWLs) for fuel tank systems to satisfy Special Federal Aviation Regulation No. 88 requirements. This proposed AD would also require the initial inspection of certain repetitive AWL inspections to phase in those inspections, and repair if necessary. This proposed AD results from a design review of the fuel tank systems. The FAA is proposing this AD to prevent the potential for ignition sources inside fuel tanks caused by latent failures, alterations, repairs, or maintenance actions, which, in combination with flammable fuel vapours, could result in a fuel tank explosion and consequent loss of the airplane.

FAA NPRM Docket Nr. FAA-2007-28385 (copy attached) proposes to mandate the Fuel System Airworthiness Limitations, including the Critical Design Configuration Control Limitations (CDCCL), for Boeing 747 series aircraft with a final compliance date of 16 December 2008. However, Boeing has voluntarily anticipated and respected the "EASA policy statement on the process for developing instructions for maintenance and inspection of fuel tank ignition source prevention" [EASA D 2005/CPRO, later revised EASA D 2006/CPRO, 8 March 2006] by publishing the Fuel System Airworthiness Limitations in Boeing Document D6-13747-CMR,

	Revision March 2006.	
	The EASA compliance time schedule is considered to be a straightforward application of Regulation (EC) No 2042/2003 Part M.A.302 in relation to the publication of the Airworthiness Limitations (including CDCCL) by the Boeing Airplane Company in document D6-13747-CMR. According to Part M.A.302(f) and (g), operators should have been aware of these Fuel Airworthiness Limitations as published by Boeing and implemented these into their aircraft maintenance programme as a result of the required annual review.	
	With this EASA AD, EASA endorses the technical content of the FAXAD 2007-XX-XX [TBD].	
	However, in deviation from the FAA AD, EASA requires the item cration to the relevant tasks into the aircraft maintenance programme before. 1 December 2007. Particular attention is requested with regard to the implementation of the required threshold inspections and grace periods, and the Aircraft Mantenano Manual (AMM), Standard Practices & Wiring Manual (SPW), and Component Maintenance Manual (CMM) references, as given in the FAA Airwe thinks a Directive, which may not have been provided in the early publication (s) of Briging document D6-13747-CMR.	
Effective Date:	[TBD: 14 days after final AD issue date	
Compliance:	EASA endorses the technical content of A AD 2007-XX-XX [TBD; currently NPRM Docket No. FAA-2007 283 5], which is attached as an appendix to this directive, except regarding paragraph (g) of that document, which must be applied as follows:	
	Maintenance Program Reviewn	
	(g) Before 01 December 2 17, reuse the approved maintenance program to incorporate the information. Section D of Revision March 2006 of Document D6-13747-CMR, except that the hitial inspections required by paragraph (h) of the FAA AD must be done of the approable compliance time specified in that paragraph. Accomplishing the register in accordance with a later revision of Document D6-13747-CMP 3 a pacceptable method of compliance if the revision is approved by EAS adjructly or to high Agency Decision 2004/04/CF by the FAA.	
Ref. Publication	Boeing Doument D6-13747-CMR, Revision March 2006 containing 747-100/200/30t. SP AWLs and CMRs. For the purposes of this AD and Document D6-747-CMR, the Model 747SR series airplane is basically a Model 747-100 series airplane with certain modifications to improve fatigue life.	
Remarks	If requested and appropriately substantiated, EASA can accept Alternative Methods of Compliance for this AD.	
	2. The closing date for comments is 28 September 2007.	
	 Enquiries regarding this AD should be referred to the AD Focal Point - Certification Directorate, EASA. E-mail: ADs@easa.europa.eu. 	
	 For any question concerning the technical content of the requirements in this AD, please contact: The Boeing Company, Fleet Support Engineering P.O. Box 3707, Seattle, Washington 98124-2207, United States of America; Telephone +1-425-237 0300; Facsimile +1-425-237 0352; E-mail csd.boecom@boeing.com; Website http://www.myboeingfleet.com. 	

§ 39.13 [Amended]

2. The Federal Aviation Administration (FAA) amends § 39.13 by adding the following new airworthiness directive (AD):

Boeing: Docket No. FAA-2007-28380; Directorate Identifier 2007-NM-088-AD.

Comments Due Date

(a) The FAA must receive comments on this AD action by August 17, 2007.

Affected ADs

(b) None.

Applicability

(c) This AD applies to the Boeing airplane models, certificated in any category, identified in the service bulletins specified Table 1 of this AD.

TABLE 1.—APPLICABILITY OF THIS AD

Boeing model—	As identified in Boeing Special Attention Service Bulletin—
747–400, 747–400D, and 747–400F series airplanes	757–26–0051, dated July 28, 2006.

Unsafe Condition

(d) This AD results from a report indicating that failure of a time delay relay on a Boeing Model 777 ELMS (electrical load management system) panel led to testing of other time delay relays at Boeing and at the supplier. Similar relays are used in the cargo fire suppression system. We are issuing this AD to ensure there is sufficient fire suppressant to control a cargo fire if the airplane is more than the relay delay time from a suitable airport, which could result in uncontrollable fire in the cargo compartment.

Compliance

(e) You are responsible for having the actions required by this AD performed within the compliance times specified, unless the actions have already been done.

Service Bulletin Reference

- (f) The term "service bulletin," as used in this AD, means the Accomplishment Instructions of the following service bulletins, as applicable:
- (1) For Model 747–400, 747–400D, and 747–400F series airplanes: Boeing Spatal Attention Service Bulletin 747–26–31, dated July 24, 2006;
- (2) For Model 757–200 series of process Boeing Special Attention Service Bulk 757–26–0051, dated July 2, 2006; and
- (3) For Model 767–200, 100, and 300F series airplanes: Boet Spect Attention Service Bulletin 767–2 0131, and July 24, 2006.

Inspection

(g) Within 24 onths e effective is AD: I date o a genera ual Sthe pot number [P/N]
201-10-14, and -6003 time delay blicab. Sthe Main Equipment t number (P/N) inspectio. TDH6103-12 relay, as applicab Center to determine in the relay was manufactured during a certain date range, in accordance with the applicable service bulletin.

Replacement

(h) Within 30 days after finding a relay manufactured during the date range specified in the service bulletin, as required by paragraph (g) of this AD: Replace the relay with a relay that was not manufactured during the specified date range, or with a relay that has been tested by the supplier and found to be unaffected by thermal expansion,

in accordance with the applicable service bulletin.

Parts Installation

(i) As of the effective date of this AD, no person may install a time delay relay, P/N TDH6103–1204, –1804, or –6003, on any airplane if the relay has a date code be compared to the code with the letter "T

Alternative Methods of Compliance (AMOCs)

- (j)(1) The Manager, Statle Aircoaft Certification Office (2.0), FAA cas the authority to approve 2.0Cs from requested in accordance is the procedures found in 14 CF 19.
- od of (2) To re est a o. ent n e or a differe compance time cedures in 14 CFR nce time complia \mathbb{Q} , follow the p for this using any a 39.19. Be proved AMOC on which th irplan AMOC applies, principal inspector Standards District notii our ap priat (PI) in FÁÂ I), or lecking a PI, your local Offi

Issued in Renton, Washington, on June 22,

Ali hrami,

Manager, Transport Airplane Directorate, Aircraft Certification Service.

[FR Doc. E7–12836 Filed 7–2–07; 8:45 am] BILLING CODE 4910–13–P

DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. FAA-2007-28385; Directorate Identifier 2006-NM-181-AD]

RIN 2120-AA64

Airworthiness Directives; Boeing Model 747–100, 747–100B, 747–100B SUD, 747–200B, 747–200C, 747–200F, 747–300, 747SR, and 747SP Series Airplanes

AGENCY: Federal Aviation Administration (FAA), Department of Transportation (DOT).

ACTION: Notice of proposed rulemaking (NF

y: The Popposes to adopt a torthine directive (AD) for all Y: 11. SUMM ew ail ing Lodel 747–100, 747–1002, B 747–200B, 747–200C, 747– 20 747SR, and 747SP serie del 747-100, 747-100B, 747-200F, 74, 200, 747SR, and 747SP series relanes. This proposed AD would revising the FAA-approved maintenance program by incorporating new airworthiness limitations (AWLs) for fuel tank systems to satisfy Special Federal Aviation Regulation No. 88 requirements. This proposed AD would also require the initial inspection of certain repetitive AWL inspections to phase in those inspections, and repair if necessary. This proposed AD results from a design review of the fuel tank systems. We are proposing this AD to prevent the potential for ignition sources inside fuel tanks caused by latent failures, alterations, repairs, or maintenance actions, which, in combination with flammable fuel vapors, could result in a fuel tank explosion and consequent loss of the airplane.

DATES: We must receive comments on this proposed AD by August 17, 2007. **ADDRESSES:** Use one of the following addresses to submit comments on this proposed AD.

- DOT Docket Web site: Go to http://dms.dot.gov and follow the instructions for sending your comments electronically.
- Government-wide rulemaking Web site: Go to http://www.regulations.gov and follow the instructions for sending your comments electronically.
- *Mail*: U.S. Department of Transportation, Docket Operations, M– 30, West Building Ground Floor, Room W12–140, 1200 New Jersey Avenue SE., Washington, DC 20590.
 - Fax: (202) 493–2251.
- Hand Delivery: Room W12–140 on the ground floor of the West Building, 1200 New Jersey Avenue SE., Washington, DC, between 9 a.m. and 5

p.m., Monday through Friday, except Federal holidays.

Contact Boeing Commercial Airplanes, P.O. Box 3707, Seattle, Washington 98124–2207, for the service information identified in this proposed AD.

FOR FURTHER INFORMATION CONTACT:

Kathrine Rask, Aerospace Engineer, Propulsion Branch, ANM-140S, FAA, Seattle Aircraft Certification Office, 1601 Lind Ave, SW., Renton, Washington 98057-3356; telephone (425) 917-6505; fax (425) 917-6590.

SUPPLEMENTARY INFORMATION:

Comments Invited

We invite you to submit any relevant written data, views, or arguments regarding this proposed AD. Send your comments to an address listed in the ADDRESSES section. Include the docket number "FAA—2007—28385; Directorate Identifier 2006—NM—181—AD" at the beginning of your comments. We specifically invite comments on the overall regulatory, economic, environmental, and energy aspects of the proposed AD. We will consider all comments received by the closing date and may amend the proposed AD in light of those comments.

We will post all comments we receive, without change, to http:// dms.dot.gov, including any personal information you provide. We will also post a report summarizing each substantive verbal contact with FAA personnel concerning this proposed A Using the search function of that **v**eb site, anyone can find and read comments in any of our do including the name of the divid who sent the comment or signed the comment on behalf of a ssociation, business, labor uni etc. You may review DOT's compl Priva Statement in the Federal Regist 11, published of 00 (65 FR ou in 19477–78), or t http:// dms. gov.

Examining 2 Do ket

You may exame, the AD docket on the Internet at http://dms.dot.gov, or in person at the Docket Operations office between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. The Docket Operations office (telephone (800) 647–5527) is located on the ground floor of the West Building at the DOT street address stated in the ADDRESSES section. Comments will be available in the AD docket shortly after the Docket Management System receives them.

Discussion

The FAA has examined the underlying safety issues involved in fuel tank explosions on several large transport airplanes, including the adequacy of existing regulations, the service history of airplanes subject to those regulations, and existing maintenance practices for fuel tank systems. As a result of those findings, we issued a regulation titled "Transport Airplane Fuel Tank System Design Review, Flammability Reduction and Maintenance and Inspection Requirements" (67 FR 23086, May 7, 2001). In addition to new airworthiness standards for transport airplanes and new maintenance requirements, this rule included Special Federal Aviation Regulation No. 88 ("SFAR 88," Amendment 21-78, and subsequent Amendments 21-82 and 21-83).

Among other actions, SFAR 88 requires certain type design (i.e., certificate (TC) and supplemental oe. certificate (STC)) holders to substan that their fuel tank systa ignition sources in the ruel ta ks. Thi requirement applies to type a sign holders for large the ine-p transport airplanes an subsequent modification bose a lanes. It ∠m to p ign reviews requires orm c and to velop desig chang s and s if their designs I tank safety nce procedu mainte the new f not me explaid in the preamble rds. A le, we ded to adopt to the ess directives to mandate any airw ages fo d necessary to address asafe conditions identified as a result these reviews.

valuating these design reviews, we have established four criteria intended to define the unsafe conditions associated with fuel tank systems that require corrective actions. The percentage of operating time during which fuel tanks are exposed to flammable conditions is one of these criteria. The other three criteria address the failure types under evaluation: single failures, single failures in combination with another latent condition(s), and in-service failure experience. For all four criteria, the evaluations included consideration of previous actions taken that may mitigate the need for further action.

We have determined that the actions identified in this proposed AD are necessary to reduce the potential of ignition sources inside fuel tanks, which, in combination with flammable fuel vapors, could result in a fuel tank explosion and consequent loss of the airplane.

Relevant Service Information

We have reviewed Boeing 747–100/ 200/300/SP Airworthiness Limitations (AWLs) and Certification Maintenance Requirements (CMRs), D6–13747-CMR, Revision March 2006 (hereafter referred to as "Revision March 2006 of Document D6-13747-CMR"). (For the purposes of Revision March 2006 of Document D6-13747-CMR, the Model 747SR series airplane basically a Model 747–100 series and lane with certain modifications to in ove fatigue life.) Section icion M. ch 2006 of Docume t De 13747-e ribes r fue. unk systems. The new AWLs new AWLs il lude:

• c. VL inspections, which are periodic as pecons of certain features for latest fances, hat could contribute to an ignition source; and

Crit al design configuration control limits of (CDCCLs), which are limitation equirements to preserve a sitical ignition source prevention team of the fuel tank system design that is necessary to prevent the occurrence of an unsafe condition. The purpose of a CDCCL is to provide instruction to retain the critical ignition source prevention feature during configuration change that may be caused by alterations, repairs, or maintenance actions. A CDCCL is not a periodic inspection.

Accomplishing the actions specified in the service information is intended to adequately address the unsafe condition.

FAA's Determination and Requirements of the Proposed AD

We have evaluated all pertinent information and identified an unsafe condition that is likely to exist or develop on other airplanes of this same type design. For this reason, we are proposing this AD, which would require revising the FAA-approved maintenance program by incorporating the information in Section D of Revision March 2006 of Document D6–13747–CMR. This proposed AD would also require the initial inspection of certain repetitive AWL inspections to phase in those inspections, and repair if necessary.

Explanation of Compliance Time

In most ADs, we adopt a compliance time allowing a specified amount of time after the AD's effective date. In this case, however, the FAA has already issued regulations that require operators to revise their maintenance/inspection programs to address fuel tank safety issues. The compliance date for these regulations is December 16, 2008. To

provide for efficient and coordinated implementation of these regulations and this proposed AD, we are using this same compliance date in this proposed AD, instead of the 18-month compliance time recommended by Boeing.

Rework Required When Implementing AWLs Into an Existing Fleet

The maintenance program revision for the fuel tank systems specified in paragraph (g) of this proposed AD, which involves incorporating the information specified in Revision March 2006 of Document D6-13747-CMR, would affect how operators maintain their airplanes. After doing that maintenance program revision, operators would need to do any maintenance on the fuel tank system as specified in the CDCCLs. Maintenance done before the maintenance program revision specified in paragraph (g) would not need to be redone in order to comply with paragraph (g). For example, the AWL that requires fuel pumps to be repaired and overhauled per an FAA-approved component maintenance manual (CMM) applies to fuel pumps repaired after the maintenance programs are revised; spare or on-wing fuel pumps do not need to be reworked. For AWLs that require repetitive inspections, the initial inspection interval (threshold) starts from the date the maintenance program revision specified in paragraph (g) is done, except as provided by paragraph (h) of this proposed AD. This proposed AD would require only the maintenance program revision specified in paragra (g), and initial inspections specifi paragraph (h). No other fleet-wi inspections need to be done

Changes to Fuel Tank System AV.

Paragraph (g) of this oposed AD would require revising to FAAe pro approved maintena m by incorporating certain forma tion D Revision specified in CMR. Paragraph (g) a acconditioning the maintenance with signal accordance with ment D6–1374 program is ision a accordance with later revisions of cument D6–13747–CMR as an acceptable method of compliance if they are approved by the Manager, Seattle Aircraft Certification Office (ACO), FAA. Paragraph (h) allows accomplishing the initial inspections and repair in accordance with later revisions of Document D6-13747-CMR as an acceptable method of compliance if they are approved by the Manager, Seattle ACO. In addition, Section D specifies that any deviations from the published AWL instructions, including AWL intervals, must be approved by the Manager, Seattle ACO. Therefore, after the maintenance program revision, any further revision to an AWL or AWL interval should be done as an AWL change, not as an alternative method of compliance (AMOC). For U.S.-registered airplanes, operators must make requests through an appropriate FAA Principal Maintenance Inspector (PMI) or Principal Avionics Inspector (PAI) for approval by the Manager, Seattle ACO. A non-U.S. operator should coordinate changes with its governing regulatory agency.

Exceptional Short-Term Extensions

Section D of Revision March 2006 of Document D6-13747-CMR has provisions for an exceptional short-term extension of 30 days. An exceptional short-term extension is an increase in an AWL interval that may be needed to cover an uncontrollable or unexp situation. For U.S.-registered airp the FAA PMI or PAI must concur v any exceptional short-term extension before it is used, unless erator identified another a ropriat procedure with the ocal reg authority. The FAA MI (Allia grant the exceptional s. t-term extension d in tion D onsultation with the Manager, withou* Seattle CO. A non-U . operator dinate charges with its uld d ning I ulatory Zency. As d in Night March 2006 of expla D6–1. 47–CMR, exceptional tensions must not be used r fleet AW⊾ extensions. An reptional short-term extension should confused with an operator's shorterm escalation authorization approved in accordance with the Operations Specifications or the operator's reliability program.

Ensuring Compliance With Fuel Tank System AWLs

Boeing has revised applicable maintenance manuals and task cards to address AWLs and to include notes about CDCCLs. Operators that do not use Boeing's revision service should revise their maintenance manuals and task cards to highlight actions tied to CDCCLs to ensure that maintenance personnel are complying with the CDCCLs. Appendix 1 of this proposed AD contains a list of Air Transport Association (ATA) sections for the revised maintenance manuals. Operators might wish to use the appendix as an aid to implement the AWLs.

Recording Compliance With Fuel Tank System AWLs

The applicable operating rules of the Federal Aviation Regulations (14 CFR parts 91, 121, 125, and 129) require operators to maintain records with the identification of the current inspection status of an airplane. Some of the AWLs contained in Section D of Revision March 2006 of Document D6-13747-CMR are inspections in which the applicable sections of the operating rules apply. Other AWLs at CDCCLs, litional which are tie ns. An maintenan e ac sting aintenance record operator's system for co ective tion is sufficient for rding a mplian with CDCCLs, g c re aj licable maintenance as lo rds identify actions manu and t hat are DCCLs.

Changes CMMs Cited in Fuel Tank System Ls

Some of the AWLs in Section D of n March 2006 of Document D6– 13747–CMR refer to specific revision levels of the CMMs as additional sources of service information for doing the AWLs. Boeing is referring to the CMMs by revision level in the applicable AWL for certain components rather than including information directly in Document D6-13747-CMR because of the volume of that information. As a result, the Manager, Seattle ACO, must approve the CMMs. Any later revision of those CMMs will be handled like a change to the AWL itself. Any use of parts (including the use of parts manufacturer approval (PMA) approved parts), methods, techniques, and practices not contained in the CMMs need to be approved by the Manager, Seattle ACO, or governing regulatory authority. For example, pump repair/overhaul manuals must be approved by the Manager, Seattle ACO.

Changes to AMMs Referenced in Fuel Tank System AWLs

In other AWLs in Section D of Revision March 2006 of Document D6-13747-CMR, the AWLs contain all the necessary data. The applicable section of the maintenance manual is usually included in the AWLs. Boeing intended this information to assist operators in maintaining the maintenance manuals. A maintenance manual change to these tasks may be made without approval by the Manager, Seattle ACO, through an appropriate FAA PMI or PAI, by the governing regulatory authority, or by using the operator's standard process for revising maintenance manuals. An acceptable change would have to maintain the information specified in

the AWL such as the pass/fail criteria or special test equipment.

Costs of Compliance

There are about 308 airplanes of the affected design in the worldwide fleet.

The following table provides the estimated costs, at an average labor rate of \$80 per hour, for U.S. operators to comply with this proposed AD.

ESTIMATED COSTS

Action	Work hours	Parts	Cost per airplane	Number of U.S registered airplanes	Fleet cost
Maintenance program revision	8 8	None		93	\$59,520 59,520

Authority for This Rulemaking

Title 49 of the United States Code specifies the FAA's authority to issue rules on aviation safety. Subtitle I, Section 106, describes the authority of the FAA Administrator. Subtitle VII, Aviation Programs, describes in more detail the scope of the Agency's authority.

We are issuing this rulemaking under the authority described in Subtitle VII, Part A, Subpart III, Section 44701, "General requirements." Under that section, Congress charges the FAA with promoting safe flight of civil aircraft in air commerce by prescribing regulations for practices, methods, and procedures the Administrator finds necessary for safety in air commerce. This regulation is within the scope of that authority because it addresses an unsafe condition that is likely to exist or develop on products identified in this rulemaking action.

Regulatory Findings

We have determined that the proposed AD would not have eralism implications under Execute Or 13132. This proposed AD would no have a substantial direc ffect on the States, on the relationship of national Government and the tween the States, or on the distribution of wer al mong responsibili e variou levels of gove me

For he reasons discuss above, I certify to the proposed regulation:

- 1. Is not a sign cant regulatory action" under Explicitly Order 12866;
- 2. Is not a "significant rule" under the DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979); and
- 3. Will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act.

We prepared a regulatory evaluation of the estimated costs to comply with this proposed AD and placed it in the AD docket. See the **ADDRESSES** section for a location to examine the regulatory evaluation.

List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Safety.

The Proposed Amendment

Accordingly, under the authorited delegated to me by the Administry or the FAA proposes to amend 14 CF part 39 as follows:

PART 39—AIRWORT NES

1. The authority continues to read as it. Ws:

Authorit 3 0. 106(, 10113, 44701.

§ 39.13 Amended]

2. The Sederal Aviation Actinists from (FAA) mends § 39.13 by acting the following new airwordness are give (AD):

Bookg: Doort No. FAA-2007-28385; Directoral Identifier 2006-NM-181-AD.

mments Due Date

The FAA must receive comments on this AD action by August 17, 2007.

Affected ADs

(b) None.

Applicability

(c) This AD applies to all Boeing Model 747–100, 747–100B, 747–100B SUD, 747–200B, 747–200F, 747–300, 747SR, and 747SP series airplanes, certificated in any category.

Note 1: This AD requires revisions to certain operator maintenance documents to include new inspections and maintenance actions. Compliance with these limitations is required by 14 CFR 43.16 and 91.403(c). For airplanes that have been previously modified, altered, or repaired in the areas addressed by these limitations, the operator may not be able to accomplish the actions described in the revisions. In this situation, to comply with 14 CFR 43.16 and 91.403(c), the operator must request approval for revision to the airworthiness limitations in the Boeing 747-100/200/300/SP Airworthiness Limitations (AWLs) and Certification Maintenance Requirements

(CMRs), D6 1747 MR, according to paragraph (g) (i) of 5 AD, as applicable.

Uns

(d) his corest is from a design review of the cell tank is to us. We are issuing this AD to purent the prential for ignition stores in ide fuel tanks caused by latent failure allevations, repairs, or maintenance actions, which, in combination with Tammable fuel vapors, could result in a fuel explosion and consequent loss of the airplant.

Compliance

(e) You are responsible for having the actions required by this AD performed within the compliance times specified, unless the actions have already been done.

Service Information Reference

(f) The term "Revision March 2006 of Document D6–13747–CMR" as used in this AD, means Boeing 747–100/200/300/SP AWLs and CMRs, D6–13747–CMR, Revision March 2006. (For the purposes of Revision March 2006 of Document D6–13747–CMR, the Model 747SR series airplane is basically a Model 747–100 series airplane with certain modifications to improve fatigue life.)

Maintenance Program Revision

(g) Before December 16, 2008, revise the FAA-approved maintenance program to incorporate the information in Section D of Revision March 2006 of Document D6–13747–CMR; except that the initial inspections required by paragraph (h) of this AD must be done at the applicable compliance time specified in that paragraph. Accomplishing the revision in accordance with a later revision of Document D6–13747–CMR is an acceptable method of compliance if the revision is approved by the Manager, Seattle Aircraft Certification Office (ACO), FAA

Initial Inspections and Repair if Necessary

(h) Do the inspections specified in Table 1 of this AD and repair any discrepancy, in accordance with Section D, "AIRWORTHINESS LIMITATIONS—SYSTEMS," of Revision March 2006 of Document D6–13747–CMR. The repair must be done before further flight. Accomplishing the actions required by this paragraph in accordance with a later revision of Document D6–13747–CMR is an acceptable method of compliance if the revision is approved by the Manager, Seattle ACO.

Note 2: For the purposes of this AD, a detailed inspection is: "An intensive examination of a specific item, installation, or assembly to detect damage, failure, or irregularity. Available lighting is normally supplemented with a direct source of good lighting at an intensity deemed appropriate.

Inspection aids such as mirror, magnifying lenses, etc., may be necessary. Surface cleaning and elaborate procedures may be required."

Note 3: For the purposes of this AD, a special detailed inspection is: "An intensive

examination of a specific item, installation, or assembly to detect damage, failure, or irregularity. The examination is likely to make extensive use of specialized inspection techniques and/or equipment. Intricate cleaning and substantial access or disassembly procedure may be required."

TABLE 1.—INITIAL INSPECTIONS

AWL No.	Description	Compliance time (whichever occurs later)			
		Threshold	Grace b od		
28-AWL-01	A detailed inspection of external wires over the center fuel tank for damaged clamps, wire chafing, and wire bundles in contact with the surface of the center fuel tank.	Before the accumulation of 36,000 total flight cycles, or within 144 months since the date of issuance of the original standard airworthiness certificate or the date of issuance of the original export certificate of airworthiness, whichever occurs first.	Within 7% ceths after e effective date of the AD.		
28-AWL-03	A special detailed inspection of the lightning shield to ground termination on the out-of-tank fuel quantity indicating system to verify functional integrity.	Before the accumulation of 36,000 total flight cycles, or within 144 to the since the date of issuance of original standard airworthiness certificate or the date of increase of the original export certification or anothiness, whichever occurs st.	ithin And other the effective ate of this D.		
28-AWL-13	A special detailed inspection of the fault current bond of the fueling shutoff valve actuator of the center wing tank to verify electrical bond.		Within 60 months after the effective date of this AD.		

Alternative Methods of Compliance (AMOCs)

(i)(1) The Manager, Seattle ACO, has the authority to approve AMOCs for this AD, if

request in accordance with the procedures found in CFR 39.19.

To request a different method of companies of different compliance time for this 10, follow the procedures in 14 CFR 39.16 decreasing any approved AMOC on

any airplane to which the AMOC applies, notify your appropriate principal inspector (PI) in the FAA Flight Standards District Office (FSDO), or lacking a PI, your local FSDO.

APPENDIX 1.—IMPLEMENTING FEL 1. 3YSTEM A. WORTHINESS LIMITATIONS ON MODEL 747–100, 747–100B, 747–100B, 747–100B, 747–200F, 747–300, 747SR, AND 747SP SERIES AIRPLANES

AWL No.	ALI/CD/	ATA action or CMM document	Task title
28-AWL-01	ALI	AMN 3-11-00/601	External Wires Over the Center Tank Inspection.
28-AWL-02 28-AWL-03	AL	SWPM -10-11	Wiring Assembly and Installation Configuration.
28-AWL-04		SWPM 20-10-15	Assembly of Shield Ground Wires.
28-AWL-05	CDCC	MM 28–22–07/401	Install the Bulkhead Fitting for the Engine Fuel Feed Tube.
28-AWL-06	CDC	AMM 29-11-22/401	Install the Heat Exchanger.
28-AV. 97	CDCCL	CMM 28–22–07, Revision 1; CMM 28–31–03, Revision 3; CMM 28–22–06, Revision 6; CMM 28–31–05, Revision 1; CMM 28–20–02, Revision 4 or Revision 9; or subsequent revisions.	
28-AWL-08	CDCCL	AMM 28–22–03/401	Install Boost Pump Housing.
		AMM 28–31–01/401	Jettison Pump Motor/Impeller Unit Installation.
28-AWL-09	ALI	AMM 28–31–00/501	Operational Test—Fault Current Detector.
28-AWL-10	CDCCL.		
28-AWL-11	CDCCL	SWPM 20–10–11	Wiring Assembly and Installation Configuration.
28-AWL-12	CDCCL	AMM 28–21–02/401	Install the Refuel Valve Control Unit.
		AMM 28–21–13/401	Install Center Wing Tank Refuel Valve Unit.
28-AWL-13	ALI	AMM 28–21–13/601	Center Wing Tank Refuel Valve—Fault Current Bond Inspection.
28-AWL-14	CDCCL	AMM 28-11-02/401	Install the Reserve and Main Tank Access Door.
		AMM 28-11-03/401	Install the Surge Tank Access Door.

APPENDIX 1.—IMPLEMENTING FUEL TANK SYSTEM AIRWORTHINESS LIMITATIONS ON MODEL 747-100, 747-100B, 747-100B SUD, 747-200B, 747-200C, 747-200F, 747-300, 747SR, AND 747SP SERIES AIRPLANES—Continued

AWL No.	ALI/CDCCL	ATA section or CMM document	Task title
28–AWL–15	CDCCL	CMM 28–40–10, Revision 13; CMM 28–40–12, Revision 9; CMM 28–40–16, Revision 1; CMM 28–40–14, Revision 8; CMM 28–40–15, Revision 0; or subsequent revisions.	
28-AWL-16	CDCCL	SWPM 20-14-12	Repair of Fuel Quantity Indicator System (FQIS) Wire Harness.
28-AWL-17	CDCCL	AMM 28–41–09/401	Replace the Fuel Tank Wire Bodle.
28–AWL–18 28–AWL–19		AMM 28–22–00/101	Engine Fuel Feed System—Trouble Scroting.

Issued in Renton, Washington, on June 22, 2007.

Ali Bahrami,

Manager, Transport Airplane Directorate, Aircraft Certification Service.

[FR Doc. E7-12815 Filed 7-2-07; 8:45 am] BILLING CODE 4910-13-P

DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. FAA-2007-28386: Directorate Identifier 2006-NM-162-AD]

RIN 2120-AA64

Airworthiness Directives; Boeing Model 747-400, -400D, and -400F Series Airplanes

AGENCY: Federal Aviation Administration (FAA), Departr t of Transportation (DOT).

ulen ring **ACTION:** Notice of propose (NPRM).

SUMMARY: The FAA prop es to adopt a ectiv new airworthiness AD) for certain Boeing Mode 47-40 -400D, rpia ¹ require and -400F § s airpla es. This proposed AL vo re revising the main e program by FAA-approved incorporating in a airwork mess limitation. (AWL) for fuel tank systems to satisfy Special deral Aviation Regulation No. 80 Aquirements. This proposed AD would also require the initial inspection of certain repetitive AWL inspections to phase in those inspections, and repair if necessary. This proposed AD results from a design review of the fuel tank systems. We are proposing this AD to prevent the potential for ignition sources inside fuel tanks caused by latent failures, alterations, repairs, or maintenance actions, which, in combination with flammable fuel vapors, could result in a

fuel tank explosion and consequent loss of the airplane.

DATES: We must receive comments on this proposed AD by August 17, 2007. **ADDRESSES:** Use one of the following addresses to submit comments on this proposed AD.

- DOT Docket Web site: Go to http://dms.dot.gov and follow the instructions for sending electronically.
- rulem • Government-wi ing We site: Go to http://www.regul and follow the instr your comments electro ally.
- Mail: rtme Dog Lation, Dog Building Gr Opt tions, M-Transpo nd Floor, Room 30, W 1200 New J W12-14 sey Avenue, SE., hingto DC 2059
 - 493-**2**51.
- v: (20. 1 Deli Room W12–140 on floor of the West Building, O New) ey Avenue, SE., (ashington, DC, between 9 a.m. and 5 , Monday through Friday, except al holidays.

Contact Boeing Commercial Airplanes, P.O. Box 3707, Seattle, Washington 98124–2207, for the service information identified in this proposed AD.

FOR FURTHER INFORMATION CONTACT:

Kathrine Rask, Aerospace Engineer, Propulsion Branch, ANM-140S, FAA, Seattle Aircraft Certification Office, 1601 Lind Ave, SW., Renton, Washington 98057-3356; telephone (425) 917-6505; fax (425) 917-6590.

SUPPLEMENTARY INFORMATION:

Comments Invited

We invite you to submit any relevant written data, views, or arguments regarding this proposed AD. Send your comments to an address listed in the **ADDRESSES** section. Include the docket number "FAA-2007-28386; Directorate Identifier 2006-NM-162-AD" at the beginning of your comments. We specifically invite comments on the

ry, eco. ¹¹ regula mic, env ntal, nd energy aspects of We will consider all the pl by the closing date comm ts recer. majo meno uno para la se comments.

set all comm mend the proposed AD in

We w st all comments we eceive, without change, to http:// Let.gov, including any personal information you provide. We will also post a report summarizing each substantive verbal contact with FAA personnel concerning this proposed AD. Using the search function of that Web site, anyone can find and read the comments in any of our dockets, including the name of the individual who sent the comment (or signed the comment on behalf of an association, business, labor union, etc.). You may review DOT's complete Privacy Act Statement in the Federal Register published on April 11, 2000 (65 FR 19477-78), or you may visit http:// dms.dot.gov.

Examining the Docket

You may examine the AD docket on the Internet at http://dms.dot.gov, or in person at the Docket Operations office between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. The Docket Operations office (telephone (800) 647-5527) is located on the ground floor of the West Building at the DOT street address stated in the ADDRESSES section. Comments will be available in the AD docket shortly after the Docket Management System receives them.

Discussion

The FAA has examined the underlying safety issues involved in fuel tank explosions on several large transport airplanes, including the adequacy of existing regulations, the service history of airplanes subject to those regulations, and existing maintenance practices for fuel tank systems. As a result of those findings, we issued a regulation titled "Transport